Experiment User Manual

Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 1 / 102 Author : J.P. Goutail

# **CONSERT Experiment User Manual**

# **ORBITER Instrument**

Experiment User Manual

Reference: RO-OCN-TN-3044Issue: 3.2Date: 09/11/04Page: 2 / 102Author : J.P. Goutail

## **DOCUMENT APPROVAL**

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# CONSERT

Orbiter

Experiment User Manual

Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 3 / 102 Author : J.P. Goutail

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CONCEDT	Reference	e: RO-OCN-TN-3044
CUNSERI	Issue	: 3.2
	Date	: 09/11/04
Orbiter	Page	: 4 / 102
Experiment User Manual	Author : .	J.P. Goutail

		5.4.3.1 to .4		17/10/02	Updated Mission tables
		5.4.3.1 to .4		08/11/02	Deltatic (= 4.95 sec) updated for all
					tables
2	8	6.1		01/09/03	TC Packet category is #12
		6.3.2.			Typing error (Packet Category for
					Memory Dump is #9)
3	0	Title	2	1/12/03	Signatures page added
3	1	6.3.1.1.	66	12/01/04	First HK rep 60 sec after Switch-on
3	2	6.3	66	09/11/04	Changed named Title
3	2	6.3.2	66	09/11/04	Service type column and Memory
					Check line integrated in table TM
					packet type summary
3	2	6.4	67	09/11/04	Added an introduction
3	2	Annexe 5	93 to	09/11/04	Changed SCET by OBT in TMs
			98		
3	2	Annexe 5	98	09/11/04	Changed & completed TM

# CONSERT

#### Orbiter

Experiment User Manual

# **TABLE OF CONTENTS**

1. General description	9
1.1. Scientific objectives	9
1.2. Experiment overview	9
2. Experiment Configuration	11
2.1. Physical	11
<ul><li>2.2. Electrical</li><li>2.2.1. OBDH Channel names and characteristics</li></ul>	11
2.3. Software	14
2.4. Budgets	14
3. Experiment operations	15
<ul> <li>3.1. Operating principles</li></ul>	15 15 16 17 19 20
3.2. Selection of Mission Table parameters for Tuning phase	22
4. Phase description	28
4.1. Phase Transition table	28
<ul> <li>4.2. Detailed phase description.</li> <li>4.2.1. Init phase</li></ul>	30 30 30 30 30 30 30 30 30
4.3. Phase budget summary	31
5. Operational procedures	32
<ul><li>5.1. Ground test sequence</li><li>5.1.1. Test Overview, definitions and short description</li></ul>	32

### Experiment User Manual

5.1.1.6. SPT (Specific Performance Test)	
5.1.1.7. Reception test (Contingency test only)	
5.1.1.8. Emission test (Contingency test only)	
5.1.1.9. Transponder test (Contingency test only)	
5.1.1.10. Instrument performance Test	
5.1.1.11. SVT (System Validation Test)	
5.1.2. Detailed Test Activities	
5.1.2.1. General Test Aspects	
5.1.3. Incoming inspection activities	
5.1.3.1. Visual inspection	
5.1.3.2. Bench Test	
5.1.4. UFT	
5.1.5. I&T Activities	
5.1.6. Unit Functional Test	
5.1.7. IST	
5.1.7.1. Connection Test	
5.1.7.2. TM & TC Verification	
5.1.7.3. Patch & Dump Test	
5.1.7.4. On Board Traffic Management (Service 3)	
5.1.7.5. Context Saving Test (Service 18)	
5.1.7.6. Science Data Transfer (Service 20)	
5.1.7.7. Private Service Verification	
5.1.7.8. Redundancy Test	
5.1.7.9. Operative Modes Verification	
5.1.7.10. Maximum Data Throughput	
5.1.7.11. Power Consumption Test	
5.1.7.12. Functional Performance Test	
5.1.8. SFT Activities	
5.1.8.1. Payload Health Check in SFT	
5.1.8.2. System Polling Test	
5.1.8.3. Payloads Parallel Operations	
5.1.8.4. Payloads Operations during Mission Simulation Tests	
5.1.9. AFT Activities	
5.1.10. SPT Activities	
5.2. On-Board Control procedures	
5.2.1. OBCP 1 : Power-On	
5.2.2. OBCP 2 : Power-Off	
5.2.3. OBCP 3 : Power-On with patch	
5.2 Elight Control Dropoduros	11
5.2.1 Switch on Conditions	
5.3.1. Switch-on Conditions	
5.3.2. List of Flight Control Flocedures	
5.3.3.1 Antenna Deployment	
5 3 3 2 Power ON for Consert	
5 3 3 3 Power OFF for Consert	۰۰۰۰۰۰ 4J ۸۸
5 3 3 4 CN-FCP-001 · Consert Orbiter Instrument Verification Test	
5 3 3 5 CN-FCP-002 · Consert Lander Instrument Verification Test	
	······································

# CONSERT<br/>OrbiterReference: RO-OCN-TN-3044<br/>Issue : 3.2<br/>Date : 09/11/04<br/>Page : 7 / 102<br/>Author : J.P. Goutail

5.3.3.6. CN-FCP-003 : Consert Instrument Solar Panel Influence	
5.3.3.7. CN-FCP-004: Consert Instrument Time Synchronization	
5.3.3.8. CN-FCP-005: Consert Instrument TC Verification	51
5.3.3.9. CN-FCP-006: Consert Instrument Patch & Dump Verification	
5.3.3.10. CN-FCP-007: Consert Orbiter Instrument Interference	53
5.3.3.11. CN-FCP-008: Consert Lander Instrument Interference	
5.3.3.12. CN-FCP-009: Consert Instrument Science Operations	
5.3.4. Mission Tables	
5.3.4.1. MTUFTO	
5.3.4.2. MTUFTL	
5.3.4.3. MTPPTO	
5.3.4.4. MIPPIL	
5.3.4.5. Alternative MIPPIO and MIPPIL	
5.5.5. DMS Monitoring	
5.4. Contingency Recovery Procedures	59
5.4.1. Redundancy concept	59
5.4.2. Software Maintenance Approach	59
6. Data Operation Handbook	60
6.1. Telecommand packet definition	
6.2. Telecommand Parameter definition	61
6.2.1. Mission Table	61
6.2.2. Direct TC	
6.2.3. Patch memory	64
6.2.4. Dump request	
6.2.5. Memory check request	
6.3. Telemetry Packet definition	
6.3.1. Data Delivery Concept	
6.3.1.1. Housekeeping reports	
6.3.1.2. Science reports (Only RTU-link)	
6.3.1.3. Event reporting	67
6.3.2. TM packet type summary	67
6.4 Telemetry Parameter definition	68
6.4.1. Successful Acknowledge	
6.4.2. Failure Acknowledge	
6.4.3. House-Keeping Report	
6.4.4. Progress Report	
6.4.5. Anomalous Report	
6.4.6. Ping Test Report	73
6.4.7. Science Report	74
6.4.8. Memory Dump Report	75
6.4.9. Memory Check Report	76
6.5. Event Packet definition	77
6.6. Context file Definition	77
6.7. Data and Dump File Definition	77

Orbiter	Date Page	: 09/11/04 : 8 / 102
Experiment User Manual	Author :	J.P. Goutail

Annex 1 : Consert Orbiter Instrument Bench Test Procedure	
Annex 2 : Consert I&T – UFT procedure for EQM	
Annex 3 : Consert IST procedure for PFM	
Annex 4 : Consert Orbiter Instrument Full Functional Test Procedure	
Annex 5 : Consert Orbiter Instrument Data Description	

Experiment User Manual

#### **1. General description**

#### **1.1. Scientific objectives**

The scientific objectives of the CONSERT experiment on the ROSETTA mission are described in the original proposal (ROSETTA MISSION : Surface Science Instruments for Champollion and RoLand, Comet Nucleus Sounding Experiment by Radiowave Transmission CONSERT, volume I, Investigation and Technical Plan) and in a recent paper published by Kofman et al., 1998. The purpose of the experiment is to determine the main dielectric properties from the propagation delay and, through modelling, to set constraints on the cometary composition (materials, porosity...) to detect large-size structures (several tens of meters) and stratification, to detect and characterize small-scale irregularities within the nucleus. A detailed analysis of the radio-waves which have passed through all or parts of the nucleus will put real constraints on the materials and on inhomogeneities and will help to identify blocks, gaps or voids. From this information we attempt to answer some fundamental questions of cometary physics : How is the nucleus built up? Is it homogeneous, layered or composed of accreted blocks (cometesimals, boulders). What is the nature of the refractory component ? Is it chondritic as generally expected or does it contain inclusions of unexpected electromagnetic properties? With the answer to these questions, it should also be possible to provide answers to the basic question of the formation of the comet. Did it form directly from unprocessed interstellar grain-mantle particles or from grains condensed in the presolar nebula? Did the accretion take place in a multi step process leading first to the formation of cometesimals which then collided to form a kilometres size body?

#### **1.2.** Experiment overview

Consert experiment consists in the rough tomography of the comet nucleus performed by the instrument (COmet Nucleus Sounding Experiment by Radiowave Transmission). It works as a time domain transponder between one module which will land on the comet surface (Lander) and another that will fly around the comet (Orbiter). The Figure 1 gives a schematic diagram of the experiment. Basically, a 90 MHz sinusoidal waveform is phase modulated by a pseudorandom code or PSK (Phase Shift Keying) Coding. Such frequency, in the radio range, is expected to minimize the losses during the propagation inside the comet material and the generated pulse code maximize the signal to noise ratio. In this experimental conditions great attempt is made on the good measurement of the mean dielectric properties and on the detection of large size embedded structures or small irregularities within the comet nucleus.

#### CONSERT ELECTRONICS ARCHITECTURE

In Figure 2, a complete structure of CONSERT experiment on the orbiter is given. At the left is the antenna which is connected to the Transmit and Receive (TR) switch. The upper part of the figure shows the receiver. From left to right, one can recognize the Radio Frequency section, with Front End Amplifier (FEA), Band Pass filters, automatic gain control (AGC), then a mixer with a 120 MHz Local Oscillator. It is followed by a wide band intermediate frequency section (WIF) at 30 MHz feeding the in-phase and quadrature detectors. A low pass filter is provided for both I and Q base band amplifiers (WBB) and a high pass section is present to eliminate DC components. Each receiver section (RF, WIF, and WBB) has a maximum gain of about 30 dB and each AGC

CONSERT	Reference Issue	e: RO-OCN-TN-3044 : 3.2
Orbiter	Date Page	: 09/11/04 : 10 / 102
Experiment User Manual	Author :	J.P. Goutail

gain take a value between 0 and -31 dB. Therefore, the total gain of the analogic part take a value between 28 and 90 dB. The in-phase and quadrature signals are converted by two 8-bits analog to digital converters. The accumulation realize in the coherent integrator systems (CANACCU) and the tuning Phase Locked Loop (PLL) will not be considered here. The bottom part of the diagram corresponds to the Transmitter with a shift register pseudo-noise (PN) generator, frequency multipliers, a phase modulator and a power amplifier.



Figure 1 : Block diagram of the CONSERT experiment. The coded signal is emitted from the Orbiter. The Lander make a coherent addition ( $\Sigma$ ) and a detection of the correlation principal peak. A clean coded signal is finally emitted with the found delay. The Orbiter accumulate the signal and send it to the earth (via the satellite interface).



Figure 2 : Electronic box of the CONSERT experiment.

#### 2. Experiment Configuration

#### 2.1. Physical

See Consert EID B vol 2.2

#### 2.2. Electrical



Figure 2.2a : Experiment Grounding Diagram





Figure 2.2b : Interconnexion Harness Bloc Diagram

For more details, see also :

Consert EID B vol 2.4 (power I/F) Consert EID B vol 2.5 (pyro I/F) Consert EID B vol 2.7 (OBDH I/F)

# CONSERT

Orbiter

Experiment User Manual

### 2.2.1. OBDH Channel names and characteristics

Function	Line	Dir.	Name	Circuit type	Connector
Serial Data Transfert	Nom	IN	NSTCT	SBDL reciever	OBDH
clock	True			HS-26C32RH	
	Nom	IN	NSTCC		OBDH
	Comp				
	Red	IN	RSTCT	SBDL reciever	OBDH
	True			HS-26C32RH	
	Red	IN	RSTCC		OBDH
	Comp				
Serial Memory Load	Nom	IN	NMLDT	SBDL reciever	OBDH
Data	True			HS-26C32RH	
	Nom	IN	NMLDC		OBDH
	Comp				
	Red	IN	RMLDT	SBDL	OBDH
	True			HS-26C32RH	
	Red	IN	RMLDC		OBDH
	Comp				
Serial Memory Load	Nom	IN	NMLST	SBDL reciever	OBDH
Sampling	True			HS-26C32RH	
	Nom	IN	NMLSC		OBDH
	Comp				
	Red	IN	RMLST	SBDL reciever	OBDH
	True			HS-26C32RH	
	Red	IN	RMLSC		OBDH
	Comp				
Serial Telemetry	Nom	IN	NSDST	SBDL reciever	OBDH
Sampling	True			HS-26C32RH	
	Nom	IN	NSDSC		OBDH
	Comp				
	Red	IN	RSDST	SBDL reciever	OBDH
	True			HS-26C32RH	
	Red	IN	RSDSC		OBDH
	Comp				
Timer Sync.	Nom	IN	NTSYT	SBDL reciever	OBDH
Pulse	True			HS-26C32RH	
	Nom	IN	NTSYC		OBDH
	Comp				
	Red	IN	RTSYT	SBDL reciever	OBDH
	True			HS-26C32RH	
	Red	IN	RTSYC		OBDH
	Comp				
Serial Telemetry	Nom	OUT	NSDDT	SBDL driver	OBDH
Data	True			HS-26C31RH	
	Nom	OUT	NSDDC		OBDH
	Comp				
	Red	OUT	RSDDT	SBDL driver	OBDH
	True			HS-26C31RH	
	Red	OUT	RSDDC		OBDH
	Comp				

#### 2.3. Software

See Consert EID B vol 2.8

#### 2.4. Budgets

See EID B Vol 2

Mission	Operation	Typical	Power (W)	ТМ	ТС	Remark
phase	type	duration		volume	volume	
Commis-	Antenna	10 min	0	0	0	Consert E-Box is off
sioning	deployment					Under SC monitoring
Commis-	Instrument	20 hours	mean 3 W	60 Mbits	10 kbit	Coordinated operations with
sioning	checkout and		peak 10 W			Consert Lander instrument
	Solar Panel					
·	Influence	0.1	2 11			
Cruise	Instrument	2 hours	mean 3 W	l Mbit	l kbit	Coordinated operations with
checkout	checkout and		peak 10 W			Consert Lander instrument
each time	Time Synchro					
it is						
possible	<b>T</b>	2.1	2 11	1.1.0.1.	0.1.1.5	
SSP	Instrument	3 hours	mean 3 W	l Mbit	2 kbit	Coordinated operations with
Lander	checkout,		peak 10 W			Consert Lander instrument
predelivery	Time Synchro					
operations	and software					
	updates	101	2 11	0.43.41.5		
SSP relay	Scientific	10 hours	mean 3 W	24 Mbits	l kbit	Coordinated operations with
(5 days)	measurements	min = 3 h	peak 10 W	(max 48		Consert Lander instrument
	set n°1	$\max = 30$		Mbits)		SPECIFIC ORBIT
	a :	h				REQUIREMENTS
SSP relay	Scientific	10 hours	mean 3 W	24 Mbits	l kbit	Coordinated operations with
(5 days)	measurements	min = 3 h	peak 10 W	(max 48		Consert Lander instrument
	set n°2	$\max = 30$		Mbits)		SPECIFIC ORBIT
	~ · · · · · ·	h				REQUIREMENTS
Rest of	Scientific	10 hours	mean 3 W	24 Mbits	l kbit	Coordinated operations with
SSP	measurements	$m_{1n} = 3 h$	peak 10 W	(max 48		Consert Lander instrument
lifetime	(values for	$\max = 30$		Mbits)		SPECIFIC ORBIT
	each comet	h				REQUIREMENTS
	scanning)					

#### 3. Experiment operations

#### **3.1. Operating principles**

#### 3.1.1. Software overview

The Consert Instrument flight software is running on an 80C32 micro-controller, with 64 K Octets of accessible RAM memory.

No real-time kernel is used, and no interrupts are used. All Activities are started on polling basis only.

The time reference inside Consert is based on an internal Clock (Sorep OCXO), which is used to start the various Consert phases and to maintain the Rosetta SCET.

The unit is a TIC. Its value is:

$$1 \text{ TIC} = 2^{14} / 10^7 = 1.6384 \text{ millisecond}$$

#### 3.1.2. Autonomy concept

After the switch-on (by a time tagged TC or spacecraft procedure OBCP), the SCET has to be delivered to the instrument (earliest 20 second, latest 60 seconds after switch on).

The Nominal/Redundant channel selection is performed automatically by the instrument based on the first transition detected on the STS (Serial Telemetry Sampling) lines after switch-on. Once a channel is selected, it remains selected until switch-off or instrument internal Reset.

After The SCET time delivery, the instrument waits for its Mission table, delivered as a private TC (data size is 20 bytes). This table defines all the timings and set-ups to be used in the instrument during the operation phase.

After the Mission table has been received, the instrument will perform all operations autonomously.

A time-tagged TC or spacecraft procedure will switch off the instrument at completion of the scientific operations.

The Rosetta Orbiter Spacecraft should be able to initiate the Consert Orbiter instrument Switch-on procedure with a time accuracy of 10 seconds with respect to ground UT.

- NB : The same request should also apply to the Consert Lander in instrument; i.e. the Lander CDMS system should be able to initiate the Consert Lander instrument Switchon procedure with a time accuracy of 10 seconds with respect to ground UT.
- In fine, the overall maximum uncertainty on the relative Consert Orbiter and Lander instruments will be less then 20 seconds.

See § 3.2. for detailed timing constraints during Consert Operation.

The complete Consert experiment is composed of :

- One Orbiter instrument (Electronics, antenna, harness)
- One Lander instrument (Electronics, antennas, harness)

Each scientific <u>measurement sequence (called scanning sequence)</u> involves the Orbiter instrument and the Lander instrument.

The duration of this Scan sequence is related to the duration of the orbit of the Rosetta Spacecraft relatively to the Lander on the rotating comet nucleus. This duration is typically of the order of one revolution around the nucleus

During this scan sequence, about 3000 individual measurements, called **<u>soundings</u>** are taken. The individual duration of this sounding is less than one second.

#### 3.1.3. Duration of one scanning sequence

This duration is typically of the order of one revolution around the nucleus and should correspond to the time when the Lander and the orbiter are separated by the comet.

During the direct line of sight periods, the synchronization occurs. This means that the duration of the data recording will not correspond to the total time of one revolution.

The number of samples is given by the following formula :

- 2 \* PI \*Radius of comet/(lambda/2)
  - where lambda is wave length

During the scanning sequence, for a circular comet with a 750m radius, about 3000 individual measurements, called soundings are taken.

This formula assumes that the rotation of S/C is much faster than the rotation of comets. In general case it will be necessary to take into accounts the relative motions in order to establish the number of samples.

#### 3.1.4. Nominal Comet and Spacecraft orbit parameters

The general structure of the CONSERT operational scenario is not dependant on the comet type that will be explored during the Rosetta mission. But a certain amount of the parameters are dependant of the shape and size of the comet nucleus and of the orbit of the spacecraft and nucleus rotation.

The numbers used here to derive the numerical parameters are :

Radius of the comet nucleus : 500 to 1500 m :	: Nominal radius = 750 m
Spacecraft orbit period around the comet :	: Minimum 3 hours : <b>Nominal : 10 hours</b> : Maximum : 30 hours

Number of CONSERT soundings during one orbit : : 3000

Note : These calculations are made for the comet Wirtanen.

for a launch in Feb-March 2004, the new target is Churyumov-Gerasimenko whose nominal radius is 2-3 times the one of Wirtanen.

<u>To keep the same surface resolution, it is necessary to increase the number of soundings with</u> the same factor (6000 to 9000 soundings). The data volume will be increased accordingly.



Orbiter

Experiment User Manual

#### 3.1.5. Operational Scenario Preparation

Information needed by the PI team before definition of measurement sequence:

- Rosetta orbital parameters around the Nucleus

- Comet shape and rotation parameters (axis, momentum)
- Lander landing site on the comet nucleus (predicted or measured)

Based on these data's, the team will compute and define the following parameters :

TON O:	Consert/Orbiter switch-on time (in UT)
T ON L :	Consert/Lander switch-on time (in UT)
TUNESTART O :	Start time for Consert/Orbiter Clock Tuning mode (in UT)
TUNESTART L :	Start time for Consert/Lander Clock Tuning mode (in UT)
SOUNDSTART :	Consert/Orbiter & Consert/Lander Sounding start time (in UT).
NBSOUNDING :	Total number of sounding performed by Consert/Orbiter & Consert/Lander
DELTASOUND :	Period between each sounding.

The Rosetta Orbiter Spacecraft should be able to initiate the Consert Orbiter instrument Switch-on time-tagged procedures with a time accuracy of 10 seconds with respect to ground UT. The Rosetta Lander Spacecraft should be able to initiate the Consert Lander instrument Switch-on time-tagged procedures with a time accuracy of 10 seconds with respect to ground UT.

#### Typical values of these numbers :

We suppose here that the soundings are made during the two third of the orbit 'behind' the comet and 5 minutes before and after this 2/3 turn.

T ON  $_{O}$  : calculated on ground, based on orbit T ON  $_{L}$  : calculated on ground, based on orbit | T ON  $_{O}$  - T ON  $_{L}$  | < 20 sec TUNESTART  $_{O}$  = T ON  $_{O}$  + 6 minutes TUNESTART  $_{L}$  = T ON  $_{L}$  + 6 minutes And : TUNESTART  $_{O}$  = TUNESTART  $_{L}$  + 21 seconds

SOUNDSTART = TUNESTART + 6 minutes

# CONSERT

Orbiter

Experiment User Manual

#### 3.1.6. Generation in the Consert EGSE of the Missioin Table TC based on these data

The complete parameter definition and Packet Structure of this TC is given in chapter 6.2 All parameters of this TC are described hereafter :

- TAB\_INDEX (8 bit unsigned value) : Index number of table, for ground reference
- TAB\_TUNETIC (32 bit unsigned value) : Time interval between instrument switch-on and start of tuning phase, expressed in TIC units, i.e. 1.6384 milliseconds periods TAB\_TUNETIC = TUNESTART o - T ON o expressed in TIC units
- TAB\_STARTTIC (32 bit unsigned value) : Time interval between instrument end of tuning phase and start of the first sounding, expressed in TIC units, i.e. 1.6384 milliseconds periods TAB STARTTIC = SOUNDSTART - TUNESTART o expressed in TIC units
- TAB\_DELTATIC (16 bit unsigned value) : Time interval between start of each, expressed in TIC units, i.e. 1.6384 milliseconds periods TAB\_DELTATIC = DELTA SOUND expressed in TIC units
- TAB\_NBSOUND (16 bit unsigned integer) : Number of soundings per scan orbit. TAB\_NBSOUND = NB SOUND
- TAB\_INITFREQ (8 bit unsigned integer) : Initial frequency of the clock (see calibration in instrument integration procedure)
- TAB\_MODEBYTE (8 bit mode byte) :
  - Only lsb is used for Orbiter
    - lsb = 0: standard value, data from FPGA (RX accumulation) are sent in TM
    - lsb = 1 : test value, tabulated value in ROM memory are sent in TM
- TAB\_MINATT (8 bit unsigned integer) : Minimum value that can be used for the sounding AGC. Can be used to force the instrument Gain to a given value. (Not operating on FM).
- TAB\_MAXATT (8 bit unsigned integer) : Maximum value that can be used for the sounding AGC. Can be used to force the instrument Gain to a given value. (Not operating on FM).

At each sounding the Attenuation value (GCW) will be constraint inside [TAB\_MINATT; TAB\_MAXATT]. (Not operating on the FM).

- TAB\_NBL\_LEVEL (8 bit unsigned integer) : Level in slow ADC unit to be reached during Tuning AGC phase on the internal connector NBL\_Level. Zero volts at input of the ADC corresponds to a value of 128
- TAB\_NBL\_ZERO (8 bit unsigned integer): Level in slow ADC unit to be detected as disappearing of the Radio Signal during Tuning phase on the internal connector NBL\_Level . Zero volts at input of the ADC corresponds to a value of 128

# CONSERT

#### Orbiter

Experiment User Manual

Mission Table	Typical value for	Increment value	Maximum value			
parameter name	AF I test					
Table Index						
TAB INDEX	1	1	255			
	Time between power	on and start of tuning				
TAB_TUNETIC	3 minutes =	1 TIC =	4294967295 TIC =			
	109863 (dec) TIC =	1.6384 msec	1954 hours =			
	00 01 AD 27 (hex) TIC		81 days			
Т	ime between end of tun	ing and Start of soundi	ng			
TAB_STARTTIC	1 minute =	1 TIC =	4294967295 TIC =			
	36621 (dec) TIC =	1.6384 msec	1954 hours =			
	00 00 8F 0D (hex) TIC		81 days			
	Time betwee	en soundings				
TAB_DELTATIC	5  seconds =	1 TIC =	65535 TIC =			
	3052 (dec) TIC =	1.6384 msec	107 seconds			
	0B EC (hex) TIC					
	Total numer	of soundings				
TAB_NBSOUND	100 (dec)	1	65535 (dec)			
	Initial clock frequency	(before start of Tuning)	· · · · · ·			
TAB INITFREQ	For FMO :	1 (dec)	255 (dec)			
	90.000 000 MHz =		× ,			
	126 (dec) =					
	7E (hex)					
	Test	Data	·			
TAB MODEBYTE	0	0	1			
Min and Max AGC attenuation						
TAB MINATT	0	1 DN	31 DN			
_		RX attenuation = $2dB$	RX attenuation = $62$ dB			
TAB MAXATT	31 (dec)	1 DN	31 DN			
_		RX attenuation $= 2dB$	RX attenuation = $62$ dB			
Minimum Level to be reached for successful tuning						
TAB NBL LEVEL	149 (dec)	1 DN	255 (dec)			
	95 (hex)		4 Volts			
Maximum Level to detect Signal disappearance						
TAB NBL ZERO	133 (dec)	1 DN	255 (dec)			
	85 (hex)		4 Volts			

In order to calculate these values easily, one can use a small Labview <sup>TM</sup> utility software that will translate Standard unit values into mission table values :

File : M\_Tab in directory Mission\_Table\_gene

NB : leading zeros are not displayed in this software, don't forget them.

#### **3.2.** Selection of Mission Table parameters for Tuning phase

The purpose of this chapter is to define the rules of Mission Table Parameter Generation in order to ensure correct Orbiter & Lander synchronization.

#### **Operational constraints :**

C1 : Switch-on time of Orbiter instrument can be defined only with an accuracy of +/- 10 seconds versus UT.

C2 : Switch-on time of Lander instrument can be defined only with an accuracy of +/-10 seconds versus UT.

#### **Tuning phase Consert constraints**

C3 : When the Tuning phase starts on Orbiter instrument (with the AGC loop), the Lander instrument should already be in radio emission (TX) phase in order to ensure good Gain convergence.

C4 : The Orbiter instrument will stop it's tuning phase even if no signal loss is detected, after a duration "Twait" after the end of the PLL loop, that is called TUNE\_TIMEOUT\_VAL in the flight software.

This duration is coded to 60 seconds (36621 TIC units) in the Flight software and can only be changed by a patch TC.

#### Software hard-coded parameters

These parameters are hard-coded in the Flight software and can only be changed by a patch TC. The definition of all these parameters are in an ASCII file called "eq\_nm\_o" (for Equivalence of Names Orbiter) and "eq\_nm\_l" (for Equivalence of Names Lander). If this file is updated the whole flight software (all modules) should be recompiled and linked.

LANDER parameter : TUNE\_DURATION : Tuning CW period is 60 seconds (36621 TIC units) NB : in the EML as delivered to DLR this value is set to only 30 seconds The value has been updated for FM (and QML)

<u>ORBITER Parameter :</u> TUNE\_TIMEOUT\_VAL = 60 seconds ( 36621 TIC units)

# CONSERT

Orbiter

Experiment User Manual

#### Parameters that can be selected by TC (in the mission table)

LANDER parameters :

Lander TUNETIC :

Duration between Lander instrument switch-on and start of the Lander Tuning phase Lander STARTTIC :

Duration between end of Tuning phase and start of the first sounding on Lander Lander INITFREQ :

Setting of the OCXO frequency (in DN value from 0 to 255). The Lander clock frequency setting will remain at this setting until instrument switch-off or direct update via a Direct\_TC.

**ORBITER** parameters :

Orbiter TUNETIC

Duration between Lander instrument switch-on and start of the Lander Tuning phase Orbiter STARTTIC :

Duration between end of Tuning phase and start of the first sounding on Orbiter. Orbiter INITFREQ :

Setting of the OCXO frequency (in DN value from 0 to 255) before the tuning phase. During the tuning phase the clock frequency will be adjusted starting from this value to try to lock on the Lander frequency. This value should correspond to a frequency as close as possible to the expected Lander frequency selected with the parameter Lander INITFREQ

Orbiter NBL\_LEVEL and Orbiter NBL\_ZERO :

Parameter used during orbiter tuning phase to adjust the Attenuation (AGC phase) and to detect the loss of signal.

Lander activities	Orbiter activities
Switch On (From Lander S/C procedure)	Switch On (From S/C OBCP procedure)
Start of Lander Tuning phase	
(Radio signal transmission TX)	
TX	Start of Orbiter Tuning phase
TX	AGC (RX Gain will increase in order to put
	the signal NBL over the value NBL_LEVEL)
TX	PLL loop
	The Orbiter clock frequency is tuned to match
	the Lander TX carrier frequency
	(duration of AGC + PLL phase is 15 sec)
TX	Wait for Radio signal loss
	(ie : wait until signal NBL goes below the
	value of NBL_ZERO)
End of TX	End of tuning phase
End of tuning phase	Internal time reference (TIC count) reset to 0
Internal time reference (TIC count) reset to 0	

#### **Tuning phase scenario**

#### Setting of Mission table parameters

As per C1 & C2 and if the requested switch-on time for Orbiter and Lander in UT are the same :

- Orbiter Switch On = Lander Switch On +/-20 sec
- C3 (Lander should start tuning phase first) gives :
- Orbiter TUNETIC > Lander TUNETIC + 20 sec.
- Let us choose : Orbiter TUNETIC = Lander TUNETIC + 21 sec.

See diagrams and worst cases on following diagrams.





#### Nominal case :

Orbiter and Lander instrument switched On at same time (UT reference time)

Duration when Radio signal is emitted by Lander before start of Orbiter AGC = 21 seconds Then Orbiter wait time for = Twait = 60 - (21 + 15) = 24 seconds





#### Worst case 1 :

Orbiter switched On 10 seconds after reference time. Lander switched On 10 seconds before reference time.

- Duration when Radio signal is emitted by Lander before start of Orbiter AGC = 21 + (10 + 10) = 41 seconds
  - Then Orbiter wait time for = Twait = 60 (41 + 15) = 4 seconds





#### Worst case 2 :

Orbiter switched On 10 seconds before reference time. Lander switched On 10 seconds after reference time.

- Duration when Radio signal is emitted by Lander before start of Orbiter AGC = 21 (10 +10) = 1 seconds
  - Then Orbiter wait time for = Twait = 60 (1 + 15) = 44 seconds

# CONSERT<br/>OrbiterReference: RO-OCN-TN-3044Issue: 3.2Date: 09/11/04Page: 28 / 102Author : J.P. Goutail

#### 4. Phase description

#### 4.1. Phase Transition table

Init phase: After switch-on and up to reception of SC Time update
Wait Mission Table phase: After *Init* and until reception of a valid mission table
Wait tuning phase: After *Wait MT* and until time for start of tuning is reached.
Tuning phase : After *Wait tuning* and until completion of tuning activities.
Wait Sounding phase : After *Tuning* and until time for start of sounding is reached.
Sounding phase : After *Wait Sounding* and until completion f the predefined n° of soundings.
End Sounding phase : After *Sounding* and until Switch-off.



Experiment User Manual

Reference: RO-OCN-TN-3044Issue: 3.2Date: 09/11/04Page: 29 / 102Author : J.P. Goutail

#### 4.2. Detailed phase description

#### 4.2.1. Init phase

After switch-on and up to reception of SC Time update

• After Hardware initialization, an event report notifying the correct initialization is generated.

Note : Time update TC (service 9.1) should be sent to the instrument after reception of this event report

#### 4.2.2. Wait mission table phase

After "Init" and until reception of a valid mission table.

- During nominal science operation, the instrument will wait until reception of a valid private TC with a Mission table (service 192.1). This table contains all the information needed for a given observation.
- In this mode (and, to be verified, only in this mode) one or more software patches (service 6.2) can be performed and memory areas can be dumped (service 6.5)
- During ground tests, direct test TC's can be sent to the instrument (service 192.2).

#### 4.2.3. Wait tuning phase

After "Wait mission table" and until time for start of tuning is reached.

#### 4.2.4. Tuning phase

After "Wait tuning" and until completion of tuning activities

- On orbiter instrument, the internal Oven Controlled Quartz Oscillator is tuned to a frequency adjusted to a radio signal received from the Lander.
- At the end of this radio signal, both Orbiter instrument and Lander instrument internal calendars are reset to zero.
- At completion of this phase, an event report is generated (either tuning success or failure).

#### 4.2.5. Wait Sounding phase

After "Tuning" and until time for start of sounding is reached.

#### 4.2.6. Sounding phase

After "Wait Sounding" and until completion of the predefined number of soundings.

- Real Science activity phase. The Science report Packets (service 20.3) are only generated in this phase.
- A Radio wave is transmitted from the Orbiter to the Lander and then back from the Lander.
- The shape and amplitude of this signal (in phase and in quadrature) are sent in the Science Report.

#### 4.2.7. End Sounding phase

After "Sounding" and until Switch-off.

		CONCEDE		Reference: RO-OCN-TN-3044	
		JNSEKI		Issue	: 3.2
	<b>Orbiter</b> Experiment User Manual			Date	: 09/11/04
				Page	: 31 / 102
				Author : J.P. Goutail	

## 4.3. Phase budget summary

Experiment phase	Power	Data Rate	Functional Use
	Usage	(Kbit/sec)	
Init	3.1 W	< 0.02	Initialization phase (duration ca 60
			seconds)
Wait mission table	2.1 W	< 0.02	Wait for the TC with Mission table
			(duration ca 60 seconds), and prepare
			mission timeline and setup.
Wait tuning	2.1 W	< 0.02	Wait for Lander tuning signal
			(duration ca 300 seconds)
Tuning	7.2 W	< 0.02	Perform internal clock tuning and
			synchronization activities (duration
			ca 300 sec)
Wait Sounding	2.1 W	< 0.02	Wait for start of sounding Mode
			(duration ca 600 sec)
Sounding	2.5 W	1.06	Perform sounding activities,
	Peak		generation of science data
	9.7 W		(duration 2 to 20 hours, comet type
			dependant)
End Sounding	2.1 W	< 0.02	Wait for switch-off

Experiment User Manual

#### 5. Operational procedures

#### 5.1. Ground test sequence

This chapter contains the inputs provided by CONSERT for the test procedure preparation of the instrument at system level on the S/C EQM and FM.

#### 5.1.1. Test Overview, definitions and short description

#### 5.1.1.1. Bench Test

The Bench Test is a replica of the bench test performed at PSR. SIS is provided by ESA and disconnected from CCS.

After reception of a specific Mission Table (MTUFT = Mission Table for Unit Functional Test), an autonomous functional test of each sub-system is performed.

#### 5.1.1.2. UFT (Unit Functional Test)

Elementary check of the experiment end-to-end data path. SCOE connected to CCS. Stimulators controlled by CCS Repeat the tests performed during Bench Test.

#### 5.1.1.3. IST (Integrated Subsystem Test)

All used services will be verified. The same Mission Table (MTUFT is used).

#### 5.1.1.4. SFT (System Functional Test)

Simulate Mission Phases and re-execute Functional Tests.

As the Consert Lander instrument is involved, coordinated operations with the Rosetta Lander are implemented and a specific Mission Table is used for this test (MTPPT).

#### 5.1.1.5. AFT (Abbreviated Functional Test)

Tests extracted from SFT.

#### 5.1.1.6. SPT (Specific Performance Test)

Part of the SFT.

This test is to verify the scientific performance. Specific test set-ups are necessary. It is composed of several sub-tests.

#### 5.1.1.7. Reception test (Contingency test only)

A dedicated radio Emitter test equipment is placed in the vicinity of the antenna and emits a signal in the 90 MHz band.

The instrument is set in a specific observation mode by Telecommand.

The signal received by the instrument is sent back via TM and analyzed in the EGSE.



#### 5.1.1.8. Emission test (Contingency test only)

A dedicated radio receiver test equipment is placed in the vicinity of the antenna and measures the signal in the 90 MHz band.

The instrument is set in a specific emission mode by Telecommand.

The signal received by the test receiver is analyzed later in the EGSE.

#### 5.1.1.9. Transponder test (Contingency test only)

Full Transponder operation test including Reception and Emission phase and a clock tuning test. A Consert Lander emulator is also used for this test.

#### 5.1.1.10. Instrument performance Test

This test is performed with both Consert/Orbiter and Consert/Lander instruments. The same Mission Table is beeing used for this test (MTPPT).

Note : During the project development, other names have been used for this test :

Clock Drift correction or Ping Pong Test.

Both systems have to be powered ON and to have a TM and TC capability at the same time. This test procedure has to be run simultaneously on the Orbiter and on the Lander, in order to perform a Complete RF Transponder sequence, including the clock correction sequence. This test is a complete Consert End-to-End test. It has to be performed on ground and also during the cruise phase.

As the Orbiter and Lander antennas will not be far away, this test can either be performed with the antennas in launch position or the antennas deployed.

#### 5.1.1.11. SVT (System Validation Test)

Closed loop test from ESOC.

#### 5.1.2. Detailed Test Activities

#### 5.1.2.1. General Test Aspects

All tests are logged. A Test Readiness Revue Report and a Post Test Revue Report is attached to the test report.

#### 5.1.3. Incoming inspection activities

#### 5.1.3.1. Visual inspection

#### 5.1.3.2. Bench Test

Bench Test Procedure

See detailed procedure in :

"Orbiter AFT Procedure", Ref. RO-OCN-PR-3043, Issue 1.0, Date 26/04/01 reproduced in "Annex 1 : Consert Orbiter Instrument Bench Test procedure"

External simulator is not necessary

- Experiment switch-on
- Wait 30 seconds
- Send SCET update
- Wait 30 seconds
- Send CON\_MISSION\_TABLE (MTAFT) (Mission table #1, short cycle)
- $-\pm 10$  minutes data acquisition (HK & SC)
- Wait for event Packet : Sounding Completed (EID 41004)
- Experiment switch-off
- Post test Data Analysis (1 hour)

SIS Configuration description

See Annex 1 : Consert Orbiter Instrument Bench Test procedure

Description of analysis tools running on SIS

See Annex 1 : Consert Orbiter Instrument Bench Test procedure

Harness connection Procedure

See Annex 1 : Consert Orbiter Instrument Bench Test procedure

#### MGSE Installation Procedure

 Deploy antenna if necessary and install antenna MGSE.
 See detailed procedure in : "Manual for handling of the CONSERT Orbiter Antenna", Ref RO-OCN-TN-3019, Issue 1.3, Date 12/06/01

#### Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 35 / 102 Author : J.P. Goutail

Experiment User Manual

#### <u>5.1.4. UFT</u>

#### 5.1.5. I&T Activities

#### Main & Redundant commands

Note : There is no internal redundancy in the instrument. The same procedure as for the Bench test is applied for each CCS configuration.

Main

- Experiment switch-on
- Wait 30 seconds
- Send SCET update
- Wait 30 seconds
- Send CON\_MISSION\_TABLE 1 (MTAFT) (Mission table #1, short cycle)
- $-\pm 10$  minutes data acquisition (HK & SC)
- Wait for event Packet : Sounding Completed (EID 41004)
- Experiment switch-off
- Post test Data Analysis (1 hour)

Redundant

- Experiment switch-on
- Wait 30 seconds
- Send SCET update
- Wait 30 seconds
- Send CON\_MISSION\_TABLE 1 (MTAFT) (Mission table #1, short cycle)
- $\pm 10$  minutes data acquisition (HK & SC)
- Wait for event Packet : Sounding Completed (EID 41004)
- Experiment switch-off
- Post test Data Analysis (1 hour)
- Data to be monitored
  - HK and SC telemetry

#### 5.1.6. Unit Functional Test

See detailed procedure in :

"Consert I&T – UFT procedure for PFM", Ref. RO-ALS-PR-4049, Issue 2d01, Date 16/05/01 and

"Consert I&T – UFT procedure for EQM", Ref. RO-ALS-PR-4049-AppB(OCN), Date 10/07/00 Partly reproduced in "Annex 2 : Consert I&T - UFT procedure for EQM"

#### Instructions to configure the SCOEs

See detailed procedures

List of commands

- Experiment switch-on
- Wait 30 seconds
- Send SCET update
- Wait 30 seconds
- Send CON\_MISSION\_TABLE 1 (MTAFT) (Mission table #1, short cycle)
- $-\pm 10$  minutes data acquisition (HK & SC)
- Wait for event Packet : Sounding Completed (EID 41004)
- Experiment switch-off
- Post test Data Analysis (1 hour)

Data to be monitored

- HK and SC telemetry

#### <u>5.1.7. IST</u>

See detailed procedure in :

"Consert IST procedure for PFM", Ref. RO-ALS-PR-4049, Issue 2d01, Date 16/05/01 and

"Consert I&T – UFT procedure for EQM", Ref. RO-ALS-PR-4049-AppA, Issue 2, Date 08/05/01 Partly reproduced in "Annex 3 : Consert IST procedure for PFM"

For information, a FFT procedure at instrument level (i.e.) using a SIS is described in : Orbiter Full Functional Test Procedure", Ref RO-OCN-TR-3042, Issue 4.2, Date 31/05/01 Reproduced in "Annex 4 : Consert Orbiter Instrument Full Functional Test Procedure"

#### OBCP List

OBCP 1 : Consert Start OBCP 2 : Consert Off OBCP 3 : Consert start with patch
Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 37 / 102 Author : J.P. Goutail

Experiment User Manual

#### 5.1.7.1. Connection Test

Service 17, 2

- Send 17,1

- Receive 17,2 : CON\_TEST\_RESP

#### 5.1.7.2. TM & TC Verification

#### Test Commands

- Experiment switch-on
- Wait 30 seconds
- Send SCET update
- Wait 30 seconds
- Send CON\_MISSION\_TABLE 1 (MTAFT) (Mission table #1, short cycle)
- $-\pm 10$  minutes data acquisition (HK & SC)
- Wait for event Packet : Sounding Completed (EID 41004)
- Experiment switch-off
- Post test Data Analysis (1 hour)

#### TM effect

Mission Table parameters are recopied in HK Telemetry.

TC generating a Service 5 Event Report

- Send CON\_MISSION\_TABLE 1 (MTAFT) (Mission table #1, short cycle)

#### 5.1.7.3. Patch & Dump Test

Memory Map

Circuit	Α	15	14	13	12	11	10	9	8	7	6	5	4	3	2	1	0	useful address
	16																	
PRGM	0	0	0	Х	Х	Х	Х	Х	Х	Х	Х	Х	Х	х	Х	Х	Х	0->3FFF
ADC/DA	0	0	1	0	0	Х	х	Х	Х	Х	0	0	0	х	Х	Х	Х	4000 (*)
С																		
LATCH1	0	0	1	0	0	Х	х	Х	Х	Х	0	0	1	х	Х	Х	Х	4010 (*)
LATCH2	0	0	1	0	0	Х	х	Х	Х	Х	0	1	0	х	Х	Х	Х	4020 (*)
FPGA	0	0	1	0	0	Х	Х	Х	Х	Х	0	1	1	х	Х	Х	Х	4030 (*)
FIFO TM	0	0	1	0	0	Х	х	Х	Х	Х	1	0	0	х	Х	Х	Х	4040 (*)
FIFO TC																		
RAM	0	0	1	0	1	х	Х	Х	х	х	х	х	х	х	х	х	Х	5000->5FFF
	0	0	1	1	0	х	Х	Х	Х	х	х	х	Х	х	Х	Х	Х	6000->6FFF
	0	0	1	1	1	Х	х	Х	Х	Х	Х	Х	Х	х	Х	Х	Х	7000->7FFF
	0	1	Х	Х	Х	Х	х	Х	Х	Х	Х	Х	Х	Х	Х	Х	Х	8000->FFFF
	1	Х	Х	Х	Х	X	Х	Х	Х	Х	Х	Х	Х	Х	Х	Х	Х	10000-> 1FFFF

#### Example of possible Data Dump

#### test A1 : Patch and Dump functions :

On SIS, perform dump of 64 words starting from address 0000 8000 (hex) see record StdD1.dat (attached) : unused memory space before patch

Send patch file patch.dat to address 0000 8000 (hex), using SIS

Perform dump of 64 words starting from address 0000 8000 (hex) see record: first 16 words are same as the patched ones.

#### Example of possible Data Load

#### test A1 : Patch and Dump functions :

On SIS, perform dump of 64 words starting from address 0000 8000 (hex) see record StdD1.dat (attached) : unused memory space before patch

Send patch file patch.dat to address 0000 8000 (hex), using SIS

Perform dump of 64 words starting from address 0000 8000 (hex) see record: first 16 words are same as the patched ones.

#### 5.1.7.4. On Board Traffic Management (Service 3)

Define HK Packets enabled

No effect. Only one type of HK Packet generated in the instrument.

#### Define HK Packets disabled

No generation of HK Packets.

#### 5.1.7.5. Context Saving Test (Service 18)

#### N/A

#### 5.1.7.6. Science Data Transfer (Service 20)

#### Define operative mode

Mission table #1

Provide RSDB

Experiment User Manual

Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 39 / 102 Author : J.P. Goutail

Preliminary RSDB inputs (Excel Sheets) are being prepared with ESOC.

#### 5.1.7.7. Private Service Verification

List of Private Services

- 192, 1 : Mission Table Update
- 192, 2 : Direct TC
- 255, 1 : Reset Telemetry Output Buffer

#### List of Private Commands

#### TM effect

#### 5.1.7.8. Redundancy Test

Define the operative mode

There is no internal redundancy in the instrument.

#### 5.1.7.9. Operative Modes Verification

**SCOE** Configuration

To be provided later

Mode Description

- Mission Table #1, short cycle.

This mode simulates the complete cycle of operation of the instrument. the Data flow is nominal (1 full SC data every 8 seconds), the total number of soundings is reduced from 3000 to a few tens to shorten the test duration.

#### 5.1.7.10. Maximum Data Throughput

#### Mode Description

Abbreviated but complete sounding sequence generated by Mission Table #1, short cycle.

Expected Data Throughput

1044 Bytes generated every 2 seconds.

Experiment User Manual

#### List of commands

CON\_MISSION\_TABLE (Mission table #1, short cycle)

#### 5.1.7.11. Power Consumption Test

Current & Voltage Monitor Description

None

Mode Description

CON\_MISSION\_TABLE (Mission table #1, short cycle).

Expected Power Consumption

See relevant chapters of ADP.

#### 5.1.7.12. Functional Performance Test

Mode Description

CON\_MISSION\_TABLE\_1 (Mission table #1, short cycle)

#### 5.1.8. SFT Activities

#### 5.1.8.1. Payload Health Check in SFT

Mode Description

CON\_MISSION\_TABLE\_2 (Mission table #2, Ping Pong Test)

#### External Stimulator Description

As the Consert Lander instrument is involved, coordinated operations with the Rosetta Lander are implemented and a specific Mission Table is used for this test (MTPPT).

#### 5.1.8.2. System Polling Test

Mode Description

CON\_MISSION\_TABLE\_2 (Mission table #2, Ping Pong Test)

Expected Data Throughput

1044 Bytes generated every 5 seconds

Experiment User Manual

#### 5.1.8.3. Payloads Parallel Operations

Applied Configuration

<u>Mode Description</u> CON\_MISSION\_TABLE\_2 (Mission table #2, Ping Pong Test)

#### 5.1.8.4. Payloads Operations during Mission Simulation Tests

Near-Sun Hibernation

Consert to be switched-on and to perform a SFT during Wake-ups

Asteroid Fly-By

Consert to be switched-on and to perform a SFT

Deep Space Hibernation Consert to be switched-on and to perform a SFT during Wake-ups

<u>Comet Observations</u> Consert to be switched-on and to perform a SFT

<u>SSP Operations</u> Consert to be switched-on and to perform a SFT in conjunction with Consert Lander.

<u>Comet Operational Phase</u> Consert to be switched-on and to perform a SFT

5.1.9. AFT Activities Same as SFT

5.1.10. SPT Activities Same as SFT

### 5.2. On-Board Control procedures

These On-Board Procedures have not been finalized yet.

<b>On Board Control</b>	Procedures (OBCP) Summary	Instrument	: CONSERT
OBCP Name	Function		Usage
OBCP 1 :	Power On the instrument		N
Consert Start	wait 30 seconds		
	Send SCET update		
	wait 30 seconds		
	Send predefined TC with mission Table		
	(CON_MISSION_TABLE)		
OBCP 2 :	After reception of Event packet : EID 41004		Ν
Consert OFF	(SOUNDING COMPLETED) : switch off Co	onsert	
OBCP 3 :	Power On the instrument		С
Consert Start with	wait 30 seconds		if a patch is needed in
patch	Send SCET update		flight
	wait 10 seconds		
	Send Memory Load (patch) TC		
	wait 10 seconds		
	Send Memory dump TC		
	wait 30 seconds		
	Send predefined TC with mission Table		
	(CON_MISSION_TABLE)		

**NB**: The time reference inside Consert is maintained by an Oven Heated Ultra Stable Quartz Oscillator (OCXO). This component is used both in Consert Orbiter and Consert Lander instruments and is needed to maintain the same radio frequency and absolute time inside both instruments for transponder operation. The Consert internal Time format is called TIC and is a sequential count of 1.6384 msec periods after the instrument-tuning phase.

All TM packets generated by Consert will be dated in SCET in CUC time format. This time will be calculated by the Consert Instrument based on the TIC value and the received SCET time update after switch-on.

#### 5.2.1. OBCP 1 : Power-On

5.2.2. OBCP 2 : Power-Off

Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 43 / 102 Author : J.P. Goutail

**Orbiter** Experiment User Manual

#### 5.2.3. OBCP 3 : Power-On with patch

OBCP Title	: PL_OBCP_5_CN.3 CONSERT POWER ON PATCH
RSDB Name	: KCNS8023
Туре	: Normal
Stored in PROM ?	: No
Remarks	:
A. Objective	

The objective of this OBCP was to switch the CONSERT instrument from the power OFF state into its initial mode and to send a S/W patch to CONSERT. The final activity is the transmission of the mission table which will cause the predefined start and performance of the CONSERT sounding mode.

Purpose of such an OBCP would be to send a patch in a defined way every time when CONSERT shall be switched on when the need for a patch once for ever would have been detected. However, as the mission table was requested to be deleted from the ON\_OBCP and to be left entirely to S/C MTL, there is no longer a meaningful need for a S/W patch OBCP as well.

The size of the S/W patch can vary between 2 bytes and 4 kB, i.e. a single patch TC is sufficient to patch CONSERT S/W. The patch TC would have to be sent to CONSERT directly before sending the mission table. As the mission table TC is to be left out, as well the patch TC (the contents of which anyhow is case dependent) can be left to S/C MTL.

Consequently the CONSERT POWER ON PATCH OBCP is identical to the normal ON\_OBCP followed by the patch and mission table TCs in the MTL. The complete patch sequence thus shall be:

- Start CONSERT with ON\_OBCP via S/C MTL
- Check successful performance of the ON\_OBCP
- Send TC "CONSERT Memory Patch" to CONSERT with acceptance report request (ZCN00602 with parameters PCND0600 thru PCND0610 and PCNG0610 thru PCNG0630)
- Wait for 1 sec (after TC has arrived at CONSERT)
- Send TC "CONSERT Memory Dump Request" to CONSERT (ZCN00605 with parameters PCND0600 thru PCND0610 and PCNG0610 thru PCNG0630)
- Send "Mission Table" TC to CONSERT with execution report request (within latest 5 min since LCL switch-on) with execution acknowledge request (ZCN19201 with parameters PCNDA011 etc. set to the actual values)
- If execution report was not sent by CONSERT resent the mission table once

Orbiter

Experiment User Manual

# **5.3. Flight Control Procedures**

#### 5.3.1. Switch-on Conditions

The only switch-on restrictions applicable for Consert are the measured temperature at the Consert E-Box TRP

Minimum Switch-on Temperature at Consert E-Box TRP : -20 °C Maximum Switch-on Temperature at Consert E-Box TRP : +50 °C Switch-on is performed by FCP : Consert-ON

#### 5.3.2. List of Flight Control Procedures

#### Instrument : CONSERT Used in all mission phases : CONSERT\_ON & CONSERT\_OFF

Mission Phase	Abbrev.	Procedure number	Procedure title
Commisionning	CVP	CONSERT_CVP_1	CN-FCP-1 : Orbiter Verification
		CONSERT_CVP_3	CN-FCP-3 : Solar Panel Influence
		CONSERT_CVP_5	CN-FCP-5 : TC Verification
		CONSERT_CVP_6	CN-FCP-6 : Patch & Dump
		CONSERT_CVP_7	CN-FCP-7 : Interference
Cruise phase 1	CR1	CONSERT_CR1_1	CN-FCP-1 : Orbiter Verification
		CONSERT_CR1_4	CN-FCP-4 : Time Synchronisation
Mars flyby	MARS	CONSERT_MARS_1	CN-FCP-1 : Orbiter Verification
		CONSERT_MARS_4	CN-FCP-4 : Time Synchronisation
Cruise phase 2	CR2	CONSERT_CR2_1	CN-FCP-1 : Orbiter Verification
		CONSERT_CR2_4	CN-FCP-4 : Time Synchronisation
Earth flyby 1	EAR1	CONSERT_EAR1_1	CN-FCP-1 : Orbiter Verification
		CONSERT_EAR1_4	CN-FCP-4 : Time Synchronisation
Cruise phase 3	CR3	CONSERT_CR3_1	CN-FCP-1 : Orbiter Verification
		CONSERT_CR3_4	CN-FCP-4 : Time Synchronisation
Aster 1 flyby	AST1	CONSERT_AST1_1	CN-FCP-1 : Orbiter Verification
		CONSERT_AST1_4	CN-FCP-4 : Time Synchronisation
Cruise phase 4	CR4	CONSERT_CR4_1	CN-FCP-1 : Orbiter Verification
		CONSERT_CR4_4	CN-FCP-4 : Time Synchronisation
Earth Flyby 2	EAR2	CONSERT_EAR2_1	CN-FCP-1 : Orbiter Verification
		CONSERT_EAR2_4	CN-FCP-4 : Time Synchronisation
Cruise phase 5	CR5	CONSERT_CR5_1	CN-FCP-1 : Orbiter Verification
		CONSERT_CR5_4	CN-FCP-4 : Time Synchronisation
Aster 2 flyby	AST2	CONSERT_AST2_1	CN-FCP-1 : Orbiter Verification
		CONSERT_AST2_4	CN-FCP-4 : Time Synchronisation

Note : Antenna deployment is not under the responsibility of Consert experiment.

Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 45 / 102 Author : J.P. Goutail

Experiment User Manual

#### 5.3.3. Detailed Procedures Inputs

#### 5.3.3.1. Antenna Deployment

Spacecraft operation

#### 5.3.3.2. Power ON for Consert

# CONSERT\_ONConsert Orbiter Instrument Switch-on (CN-FCP-000)Procedure Objective :This procedure performs the power-on sequence for Consert. The instrument will select

autonomously the active TM/TC channel and the active power channel.

Step n°	Time from Consert LCL	Activity	TM/TC packet	TM/TC parameter
	(hh:mm:ss)			
1	00:00:00	LCL ON		
2	00:00:05	Verify Event : "Init OK" received	YCN0D501	NCNA0505 = 41001
	00:00:10	Check Parameters : LCL Voltage LCL Current		
	00:00:20	Verify first HK packet : Initialisation Performed OCXO Temperature DIGI Board Temperature	YCN00325	NCND0331 = 1 NCND0339 = ? °C NCND0341 = ? °C
3	00:00:30	If necessary, Send Patch Parameters	ZCN00602	
4	00:00:40	If necessary, Dump Memory for Patch verification	ZCN00605	
5	00:00:50	Send SCET update to Consert	ZCN00901	
6	00:01:00	Verify HK packet : LOBT received	YCN00325	NCND0338 = 1



#### 5.3.3.3. Power OFF for Consert

CONSERT	OFF Consert Orbiter Instrument Switch-Off				
	Can be used fo	r emergency switch o	off.		
Procedure	<b>Objective :</b>				
This proced	lure performs the power-off	sequence for Consert.	It can be run at any i	noment, no	
specific safe	e procedure in needed prior t	to Consert Off.			
No specific	recovery procedure is neede	ed in case of an emerge	ency Switch-Off.		
Step n°	Time from Consert LCL	Activity	TM/TC packet	TM/TC	
	switch On			parameter	
	(hh:mm:ss)				
1	00:00:00	LCL OFF			
2	00:00:10	00:00:10 Check Parameters :			
	LCL Voltage				
		LCL Current			

#### 5.3.3.4. CN-FCP-001 : Consert Orbiter Instrument Verification Test

CN-FCP-001 Consert Orbiter Instrument Verification Test	
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**Procedure Objective :** 

This procedure performs an autonomous test of all functions of the Consert Orbiter instrument. The ambient noise level is also measured.

Step n°	Time from Consert LCL	Activity	TM/TC nacket	TM/TC parameter
	switch On		расксі	
	(hh:mm:ss)			
1	00:00:00	Run FCP		
		CONSERT_ON		
2	00:02:00	Send Consert Orbiter	ZCN19201	after this TC the instrument
		Mission Table :		will perform autonomous
		MTUFTO		activities during 18 minutes
3	00:02:10	Verify HK packet :	YCN00325	
		"Mission Table		NCND0332 = 1
		received"		
4	00:05:30	Verify Event :	YCN0A501	
		"Tuning ok" received		NCNA0510 = 41002
5	00:05:40	Verify HK packet	YCN00325	
		"Tuning Phase		NCND0333 = 1
		Performed"		
6	00:06:30	Verify Event :	YCN0B501	
		"Sounding Started"		NCNA0520 = 41003
		received		
7	00:06:40	Verify HK packet :	YCN00325	
		"Sounding		NCND0334 = 1
		Continuing"		
8	00:18:00	Verify Event	YCN0C501	
		"Sounding Completed"		NCNA0530 = 41004
		received		
9	00:18:15	Run FCP		
		CONSERT OFF		

#### 5.3.3.5. CN-FCP-002 : Consert Lander Instrument Verification Test

#### CN-FCP-002 Consert Lander Instrument Verification Test

#### **Procedure Objective :**

This procedure performs an autonomous test of all functions of the Consert Lander instrument. The ambient noise level is also measured.

It can be performed with the antenna either folded or deployed.

#### See Consert Lander Instrument User Manual for details.

Step n°	Time from Consert LCL switch On (hh:mm:ss)	Activity	TM/TC packet	TM/TC parameter
1	00:00:00	Switch-on Consert		
		Lander		
2	00:02:00	Send Consert		After this TC the instrument
		Lander Mission		will perform autonomous
		Table :		activities during 15 minutes
		MTUFTL		
3	00:18:15	Switch-off Consert		
		Lander		

#### Orbiter

Experiment User Manual

#### 5.3.3.6. CN-FCP-003 : Consert Instrument Solar Panel Influence

#### CN-FCP-003 Consert Instrument Solar Panel Influence

#### **Procedure Objective :**

This procedure measures the influence of the Solar Arrays position versus the clock correction and gain and calibration parameters.

The measurements need to be made in steps of every 3-10 degrees of the angle between the S/C body and Solar Arrays.

Moving to the next step and stopping is preferable to a continuous very slow drift.

This procedure needs the Orbiter and Lander part of CONSERT to be switched-on.

It must be performed with the Consert Orbiter antenna deployed.

Step n°	Time from	Activity	TM/TC	TM/TC parameter
	Consert LCL		packet	
	switch On			
	(hh:mm:ss)			
1	00:00:00	Run FCP		Note :
		CONSERT_ON		The Lander instrument should
				be switched on at the same
				moment, see chapter 3.1.2
2	00:02:00	Send Consert Orbiter	ZCN19201	After this TC the instrument
		Mission Table :		will perform autonomous
		МТРРТО		activities during 15 minutes
3	00:17:00	Verify Event	YCN0C501	
		"Sounding Completed"		NCNA0530 = 41004
		received		
4	01:05:15	Run FCP		
		CONSERT_OFF		
5		Rotate Solar Arrays 3-		
		10 degrees		
6		If complete rotation of		
		Solar Arrays less than		
		180°, Go to step 1		

#### 5.3.3.7. CN-FCP-004: Consert Instrument Time Synchronization

#### CN-FCP-004 Consert Instrument Time Synchronisation

#### **Procedure Objective :**

This procedure measures the Clock Drift between the Orbiter and Lander instrument and allows to correct excessive drifts during the cruise.

After Tuning phase has been performed, some TCs are sent to change the OCXO frequency. It is done to verify the signal amplitude variation as a function of the frequency.

It also checks the overall health of the whole Consert experiment (Orbiter + Lander).

Step n°	Time from	Activity	TM/TC	TM/TC parameter
	Consert LCL		packet	
	switch On			
	(hh:mm:ss)			
1	00:00:00	Run FCP		Note :
		CONSERT_ON		The Lander instrument should
				be switched on at the same
				moment, see chapter 3.1.2
2	00:02:00	Send Consert Orbiter	ZCN19201	After this TC the instrument
		Mission Table :		will perform autonomous
		МТРРТО		activities during 18 minutes
3		Verify Event :	YCN0B501	
		"Sounding Started"		NCNA0520 = 41003
		received		
4	00:50:00	Change clock setting	ZCN19202	This TC may be repeated 15
		value to measure the		times to select various clock
		fine clock offset		settings around the value set
				by the tuning.
5	00:51:00	Change clock setting	ZCN19202	
		value to measure the		
		fine clock offset		
6	00:17:00	Verify Event	YCN0C501	
		"Sounding Completed"		NCNA0530 = 41004
		received		
7	01:05:15	Run FCP		
		CONSERT OFF		



#### 5.3.3.8. CN-FCP-005: Consert Instrument TC Verification

#### CN-FCP-005 Consert Instrument TC Verification

**Procedure Objective :** 

This procedure verifies the capability of the instrument to accept and execute Direct Commands. It also checks the overall health of the Consert Orbiter instrument.

The same kind of procedure may be executed for the Lander instrument.

Step n°	Time from Consert	Activity	TM/TC	TM/TC parameter
	LCL switch On		packet	
	(hh:mm:ss)			
1	00:00:00	Run FCP		
		CONSERT_ON		
2	00:02:00	Send Consert	ZCN19201	after this TC the instrument
		Orbiter Mission		will perform autonomous
		Table :		activities during 15 minutes
		MTUFTO		
3	00:50:00	Change clock	ZCN19202	This TC may be repeated 15
		setting value to		times to select various clock
		measure the fine		settings around the value set
		clock offset		by the tuning.
4	00:51:00	Change clock	ZCN19202	
		setting value to		
		measure the fine		
		clock offset		
5	00:17:00	Verify Event	YCN0C501	
		"Sounding		NCNA0530 = 41004
		Completed"		
		received		
6	01:05:15	Run FCP		
		CONSERT OFF		

#### Orbiter

Experiment User Manual

#### 5.3.3.9. CN-FCP-006: Consert Instrument Patch & Dump Verification

#### CN-FCP-006 Consert Instrument Patch & Dump Verification

#### **Procedure Objective :**

This procedure verifies the capability of the instrument to accept software patches and to dump its memory.

It also checks the overall health of the Consert Orbiter instrument.

The same kind of procedure may be executed for the Lander instrument.

Step n <sup>o</sup>	Time from Consert	Activity	TM/TC	TM/TC parameter
-	LCL switch On	v	packet	Ĩ.
	(hh:mm:ss)		-	
1	00:00:00	Run FCP		
		CONSERT_ON		
2	00:00:30	Send Patch	ZCN00602	0xAAAA @ 0x8000
		Parameters		
3	00:00:40	Dump Memory for	ZCN00605	
		Patch verification		
4	00:02:00	Send Consert	ZCN19201	After this TC the instrument
		Mission Table :		will perform autonomous
		MTUFTO		activities during 15 minutes
5	00:17:00	Verify Event	YCN0C501	
		"Sounding		NCNA0530 = 41004
		Completed"		
		received		
6	01:05:15	Run FCP		
		CONSERT_OFF		

#### Orbiter

Experiment User Manual

#### 5.3.3.10. CN-FCP-007: Consert Orbiter Instrument Interference

#### CN-FCP-007 Consert Orbiter Instrument Interference

#### **Procedure Objective :**

This procedure measures the interferences between the Consert Orbiter instrument and other subsystems of the Spacecraft.

It also checks the overall health of the Consert Orbiter instrument.

The same kind of procedure may be executed for the Lander instrument.

	1	11		1
Step n°	Time from Consert	Activity	TM/TC	TM/TC parameter
	LCL switch On		packet	
	(hh:mm:ss)			
1	00:00:00	Run FCP		
		CONSERT_ON		
2	00:02:00	Send Consert	ZCN19201	After this TC the instrument
		Orbiter Mission		will perform autonomous
		Table :		activities during 15 minutes
		MTUFTO		
3		Wait for the end of		
		Interference		
		operations		
4	01:05:15	Run FCP		
		CONSERT_OFF		

#### 5.3.3.11. CN-FCP-008: Consert Lander Instrument Interference

#### CN-FCP-008 Consert Instrument Interference Lander

#### **Procedure Objective :**

This procedure measures the interferences between the Consert Lander instrument and other subsystems of the Spacecraft and Lander.

It also checks the overall health of the Consert Lander instrument.

The same kind of procedure may be executed for the Orbiter instrument.

It can be performed with the antennas either folded or deployed.

#### See Consert Lander Instrument User Manual for details.

Step n°	Time from Consert LCL switch On (hh:mm:ss)	Activity	TM/TC packet	TM/TC parameter
1	00:00:00	Switch-on Consert		
		Lander		
2	00:02:00	Send Consert		After this TC the instrument
		Lander Mission		will perform autonomous
		Table :		activities during 15 minutes
		MTUFTL		
3	00:18:15	Switch-off Consert		
		Lander		

# Orbiter

Experiment User Manual

# 5.3.3.12. CN-FCP-009: Consert Instrument Science Operations

<b>CN-FCP-</b>	009 Conser	t Instrument Science	e Mode	
Procedure Objective : Science Operations It must be performed with the antennas deployed.				
Step n°	Time from Consert LCL switch On (hh:mm:ss)	Activity	TM/TC packet	TM/TC parameter
1	00:00:00	Run FCP CONSERT_ON		Note : The Lander instrument should be switched on at the same moment, see chapter 3.1.2
2	00:02:00	Send Consert Mission Table : MTSCO	ZCN19201	After this TC the instrument will perform autonomous activities during 10 hours
3	01:05:00	Verify Event "Sounding Finished" received	YCN0C501	NCNA0 = 41004
4	01:05:15	Run FCP CONSERT_OFF		

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#### Experiment User Manual

#### 5.3.4. Mission Tables

Note : 1 Tic = 1.6384 msec

#### 5.3.4.1. MTUFTO

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MTUI	MTUFTO : Orbiter Mission Table for Functional Test (10 16bits Words)			
Parameter #	TC Data Word (Hex)	Signification		
PCNGA010	0100	Mission table index & Spare		
PCNGA020	00038C60	TUNETIC = 232544 Tics (381 seconds)		
PCNGA030	00008FOD	STARTTIC = 36621 Tics (60 seconds)		
PCNGA040	0BCD	DELTATIC = 3021 Tics (4.95 seconds)		
PCNGA050	0078	NBSOUNDING (= 120)		
PCNGA060	8000	INIT FREQ =128 & Mode byte setting		
PCNGA070	001F	MIN ATT = 0 & MAX ATT = 31		
PCNGA080	9585	NBL Level = 149 & NBL zero = 133		

Total duration of this mode : circa 16 minutes

#### 5.3.4.2. MTUFTL

MTUFTL : Lander Mission Table for Functional Test (10 16bits Words)			
Parameter #	TC Data Word (Hex)	Signification	
1	0301	Mission table indicator & table index	
2	0003	TUNETIC (B3 & B2) = 219727 Tics (360 seconds)	
3	5A4F	TUNETIC (B1 & B0)	
4	0000	STARTTIC (B3 & B2) = 36621 Tics (60 seconds)	
5	8F0D	STARTTIC (B1 & B0)	
6	0BCD	DELTATIC = 3021 Tics (4.955 seconds)	
7	0064	NBSOUNDING = 100	
8	8305	INIT FREQ (=131) & FIOW RATIO (=5)	
9	0000	MODE BYTE $(= 0)$ & MIN ATT $(= 0)$	
10	1F00	MAX ATT (= 0) & PAD Field (=0)	

Orbiter

Experiment User Manual

Note : 1 Tic = 1.6384 msec

#### 5.3.4.3. MTPPTO

MTP	MTPPTO : Orbiter Mission Table for Ping Pong Test (10 16bits Words)			
Parameter #	TC Data Word (Hex)	Signification		
PCNGA010	0100	Mission table index & Spare		
PCNGA020	00038C60	TUNETIC = 232544 Tics (381 seconds)		
PCNGA030	00008FOD	STARTTIC = 36621 Tics (60 seconds)		
PCNGA040	0BCD	DELTATIC = 3021 Tics (4.95 seconds)		
PCNGA050	0078	NBSOUNDING (= 120)		
PCNGA060	8000	INIT FREQ =128 & Mode byte setting		
PCNGA070	001F	MIN ATT = 0 & MAX ATT = 31		
PCNGA080	9585	NBL Level = 149 & NBL zero = 133		

Total duration of this mode : circa 18 minutes

#### 5.3.4.4. MTPPTL

МТР	MTPPTL : Lander Mission Table for PingPong Test (10 16bits Words)			
Parameter #	TC Data Word (Hex)	Signification		
1	0301	Mission table indicator & table index		
2	0003	TUNETIC (B3 & B2) = 219727 Tics (360 seconds)		
3	5A4F	TUNETIC (B1 & B0)		
4	0000	STARTTIC(B3 & B2) = 36621 Tics (60 seconds)		
5	8F0D	STARTTIC(B1 & B0)		
6	0BCD	DELTATIC = 3021 Tics (4.95 seconds)		
7	0064	NBSOUNDING = 100		
8	8305	INIT FREQ (=131) & FIOW RATIO (=5)		
9	0000	MODE BYTE $(= 0)$ & MIN ATT $(= 0)$		
10	1F00	MAX ATT (= 0) & PAD Field (=0)		

#### 5.3.4.5. Alternative MTPPTO and MTPPTL

The monitoring of the experiment primary current is done by the spacecraft every second. During the IST, it showed only standby power (90 mA), sampling regularly between the soundings.

A way to monitor the complete power consumption of the instrument is to read the power consumption with a repetition period that is not a fraction of the sounding period of the instrument.

We propose to slightly change the parameter that controls the sounding repetition time in the instrument (from 5 seconds to 4.95 seconds).

This modification is validated and included in the mission tables (§ 5.4.3.1 to .4)

Orl	Orbiter Mission Table modification for primary current monitoring				
Parameter #	TC Data Word (Hex)	Signification			
PCNGA040	0BCD	DELTATIC = 3021 Tics (4.95 seconds)			

#### Orbiter Tables

#### Corresponding Lander Tables

Lander Miss	Lander Mission Table modification to match the Orbiter primary current monitoring			
Parameter #	TC Data Word (Hex)	Signification		
	0301			
6	0BCD	DELTATIC = 3021 Tics (4.95 seconds)		
	0064			

#### 5.3.5. DMS Monitoring

No in flight parameter monitor is requested for Consert. The instrument should be switched off at completion of its sounding period (notified by an event).

DMS Monitoring	Instrument : CONSER	Γ
Monitored Entity	Monitoring requirements	Action on
		Event
Parameters		
None	no parameter monitoring	
	requirements	
Events		
Event 1 :	at completion of sounding activities,	Switch OFF the
EID = 41004 (SOUNDING COMPLETED)		instrument

# 5.4. Contingency Recovery Procedures

#### 5.4.1. Redundancy concept

Except at interface level (power and OBDH), there is no internal redundancy inside Consert.

The Nominal/Redundant channel selection is performed automatically by the instrument based on the first transition detected on the STS (Serial Telemetry Sampling) lines after switch-on. Once a channel is selected, it remains selected until switch-off or instrument internal Reset.

In case of a Failure of an Interface, the instrument has to by switched off, then the SC interface channel has to be switched to the alternate. Then the Consert instrument can be switched On again.

#### 5.4.2. Software Maintenance Approach

The Consert Flight Software is stored in a Read Only Memory (27C256 TRBDB-12, 32kOctets), programmed on ground and soldered on the Consert Digital Board before instrument delivery. After each instrument switch-on, the content of this ROM is loaded in the RAM memory and the software is executed there.

Thus, software patching of all parts of the software is possible during flight. After each switch-off or Reset, the patches are lost.

As the Consert flight software is very simple and does not contain any data analysis, only very limited patching and dumping of memory areas may be done to verify the correct implementation of the patch and the integrity of the software in case of problems.

# 6. Data Operation Handbook

# 6.1. Telecommand packet definition

Sub	Database TC	Service Request (TC)	Usage / remark			
Гуре	Name					
	SERVICE 3					
5	ZCN00305	Enable HK Report				
6	ZCN00306	Disable HK Report	Not to be used in nominal obs.			
		SERVICE 6				
2	ZCN00602	Load Memory by absolute add				
5	ZCN00605	Memory Dump request by				
		absolute add				
9	ZCN00609	Memory check request by				
		absolute add				
		SERVICE 9				
1	ZCN00901	Accept time update	To be sent only once per			
			scanning sequence, after			
			instrument switch-on			
		SERVICE 17				
1	ZCN01701	Connection Test Request				
		SERVICE 20				
1	ZCN02001	Enable Science Report (RTU)				
2	ZCN02002	Disable Science Report (RTU)	Not to be used in nominal obs.			
		SERVICE 192				
1	ZCN19201	Mission table Uplink	To be sent only once per			
		_	scanning sequence, after			
			instrument switch-on and time			
			update			
2	ZCN19202	Direct TC	For test purposes only			
		SERVICE 255				
1	ZCN25501	Reset TM buffer				

Remark : Packet category is #12 for all TCs

Orbiter

Experiment User Manual

# 6.2. Telecommand Parameter definition

Alternate format tables of this information can be found in Annex 5.

#### 6.2.1. Mission Table

<b>Telecommand Packet Information</b>		ition	ZCN19201
Packet Name	CON_MIS	SION_TABLE	Instrument : Consert
Packet Function	Consert M	ission table uplink TC	
Generation Rules	to be sent a	after instrument switch-o	on
<b>Header Informatio</b>	n		
Process ID	59	Packet Category	12
Service Type	192	Service Subtype	1
Structure ID		Packet Length	Datafield $= 26$
		in Bytes	Total packet length = $26 + 6 = 32$
Data Field Informa	<u>ation</u>		
Data Field	Field	Remark	
	Structure		
Data field Header	4 Bytes	as per EID A	
PCNGA010	2 Bytes	TAB_INDEX : mission	n table index
PCNGA020	4 Bytes	TAB_TUNETIC B3	to B0 (MSB to lsb) Start of tuning
PCNGA030	4 Bytes	TAB_STARTTIC B3	to B0 (MSB to lsb) Start of Sounding
PCNGA040	2 Bytes	TAB_DELTATIC B1	; B0 (MSB ; lsb) Time step in TIC
PCNGA050	2 Bytes	TAB_NBSOUND B1;	B0 : Total number of soundings
PCNDA061	1 Byte	TAB_INITFREQ : cloc	ck initial frequency setting
PCNDA062	1 Byte	TAB_MODE : mode E	Byte setting
PCNDA071	1 Byte	TAB_MINATT : minin	mum GCW (attenuation)
PCNDA072	1 Byte	TAB_MAXATT : max	timum GCW (attenuation)
PCNDA081	1 Byte	TAB_NBL_LEVEL : 1	evel to be reached during NBL AGC
PCNDA082	1 Byte	TAB_NBL_ZERO : le	vel to be detected during NBL zero
		detection	
Packet Error	2 Bytes	not tested by Consert	
Control			
Notes :			

Mode Byte bit pattern definition :

Bit 0 (lsb) = DataSource 0 = FPGA 1 = Simulated data Bit 1 to Bit 7 : not used

# 6.2.2. Direct TC

<b>Telecommand Packet Information</b>			formation		ZCN19202		
Packet Name	(	CON	L_DIRECT	TC	Instrument : Consert		
Packet Function	on (	Cons	sert direct 7	ГС			
Generation Ru	les 7	To b	e used for t	tests only			
Header Inform	nation			· · · · · · · · · · · · · · · · · · ·			
Process ID	5	59		Packet Category	12		
Service Type	1	192		Service Subtype	2		
Structure ID			Packet Length		datafield length $= 8$		
			in Bytes Total packet length = $8 + 6 =$				
Data Field Inf	formati	ion	_				
Data Field	I	Field	d	Remark			
	S	Struc	cture				
Data field Hea	der 4	4 By	tes	as per EID A			
PCNDB011	1	1 By	te	DIR_COMMAND : see	e note 4		
PCNDB012	1	I By	te	DIR_PARAM : see not	e 4		
Packet Error	2	2 By	tes	not tested by Consert			
Control							
Notes :							
Note 4 :	DID				<b></b>		
DIR		<b>K</b>		Action	Remarks		
COMMAND	PAKA	A IVI	Q = 4 = 1 = -1 = 1				
5	X		Set Clock I	DAC to X			
6	0		CLEAK I	XPON			
6	l		SET TXPO	JN			
7	0		CLEAR R	XPON			
7	1		SET RXP	N			
8	0		CLEAR T	RCOM			
8	1		SET TRC	OM			
9	0		CLEAR T	UNING COM			
9	1		SET TUN	ING COM			
А	0		CLEAR T	RPON			
А	1		SET TRPO	DN			
В	0		SWITCHS	SEO OFF	MESCOM is OFF		
B	1		SWITCHS	SEQ ON	FPGA is in reset state all time		
	1				MESCOM is ON		
E	x		set Gain ((	GCW) to X			
F	0		Set RVPA	SS OFF (measurement)			
F	1		Set RVPA	SS ON (Tuning)			
10	$\frac{1}{n(0 t_{0})}$	2)	Set codo o	$\frac{1}{1} \frac{1}{1} \frac{1}$	0 = Code from EDGA Nothing if		
10		, 2)	(I ander O	nlv)	$FPGA \cap FF$		
				111 <i>y J</i>	1 = Delta  312		
					2 = CW (sinus)		

Experiment User Manual

Reference: RO-OCN-TN-3044Issue: 3.2Date: 09/11/04Page: 63 / 102Author : J.P. Goutail

CONCEDT	Reference:	RO-OCN-TN-3044
CONSERI	Issue	: 3.2
	Date	: 09/11/04
Orbiter	Page	: 64 / 102
Experiment User Manual	Author : J.P	. Goutail

# 6.2.3. Patch memory

<b>Telecommand Pac</b>	ket Information	1	ZCN00602		
Packet Name	CON_MEMO	РАТСН	Instrument : Consert		
Packet Function	Consert Memo	bry Patch TC			
Generation Rules	To be used to	update flight software			
Header Informatio	n				
Process ID	59	Packet Category	12		
Service Type	6	Service Subtype	2		
Structure ID		Packet Length	$12 + 2*block_length$		
		in Bytes	$6 + 12 + 2$ *block_length		
<b>Data Field Informa</b>	ation				
Data Field	Field	Remark			
	Structure				
Data field Header	4 Bytes	as per EID A			
PCND0600	1 Byte	Always 60 (dec) for Consert Memory			
Memory_ID					
PCND0610	1 Byte	Always 1 for Consert	t		
number of blocks					
PCNG0610	4 Bytes	MSB to lsb , the two f	irst bytes are always 00 (only64k		
Start Address		octet mem)			
PCNG0620	2 Bytes	MSB lsb, length in 16	bit words $=$ n		
Block length = $n$					
PCNG0630	2n Bytes	Data to be written in n	nemory		
Data					
Packet Error	2 Bytes	not tested by Consert			
Control					
Notes : Only one va	alid Memory II	) for Consert = 60 dec			
Only one memory	Only one memory segment can be dumped at each time				

CONCEDE	Reference	e: RO-OCN-TN-3044
CONSERI	Issue	: 3.2
	Date	: 09/11/04
Orbiter	Page	: 65 / 102
Experiment User Manual	Author : J	J.P. Goutail

# 6.2.4. Dump request

<b>Telecommand Pack</b>	ket Information	l	ZCN00605	
Packet Name	CON_MEMO_	DUMP_RQ	Instrument : Consert	
Packet Function	Consert Memo	ry Dump request TC		
Generation Rules	To be used to	request the dump of a pa	art of the memory	
<b>Header Informatio</b>	n			
Process ID	59	Packet Category	12	
Service Type	6	Service Subtype	5	
Structure ID		Packet Length	Datafield = 14	
		in Bytes	Total packet length = 20	
<b>Data Field Informa</b>	ntion			
Data Field	Field	Remark		
	Structure			
Data field Header	4 Bytes	as per EID A		
PCND0600	1 Byte	Always 60 (dec) for Consert Memory		
Memory_ID				
PCND0610	1 Byte	Always 1 for Consert		
number of blocks				
PCNG0610	4 Bytes	MSB to lsb, the two f	irst bytes are always 00 (only 64k	
Start Address		octet mem)		
PCNG0620	2 Bytes	MSB lsb, length in 16	bit words $=$ n	
Block length = $n$				
Packet Error	2 Bytes	not tested by Consert		
Control				
Notes : Only one va	alid Memor <mark>y ID</mark>	for Consert = $60 \text{ dec}$		
Only one memory s	segment can be	dumped at each time		
Will generate a CO	N_MEMO_DU	MP_TM		

CONCEDT	Reference: I	RO-OCN-TN-3044
CUNSERI	Issue	: 3.2
	Date	: 09/11/04
Orbiter	Page	: 66 / 102
Experiment User Manual	Author : J.P	. Goutail

# 6.2.5. Memory check request

<b>Telecommand Pac</b>	ket Information	1	ZCN00609
Packet Name	CON_MEMO	_CHECK_RQ	Instrument : Consert
Packet Function	Consert Memo	bry check request TC	
Generation Rules	To be used to	request the calculation of	of the CRC check sum of a part of the
	memory	-	
Header Informatio	n		
Process ID	59	Packet Category	12
Service Type	6	Service Subtype	9
Structure ID		Packet Length	Datafield = 14
		in Bytes	Total packet length $= 20$
<b>Data Field Informa</b>	ation		
Data Field	Field	Remark	
	Structure		
Data field Header	4 Bytes	as per EID A	
	1 Byte	Always 60 (dec) for C	Consert Memory
Memory_ID			
	1 Byte	Always 1 for Consert	;
number of blocks			
	4 Bytes	MSB to lsb, the two f	irst bytes are always 00 (only 64k
Start Address		octet mem)	
	2 Bytes	MSB lsb, length in 16	bit words $=$ n
Block length $=$ n			
Packet Error	2 Bytes	not tested by Consert	
Control			
Notes : Only one va	alid Memory II	) for Consert = 60 dec	
Only one memory	segment can be	dumped at each time	
Will generate a CC	ON_MEMO_CH	IECK_TM	

### 6.3. Telemetry Packet definition

#### 6.3.1. Data Delivery Concept

#### 6.3.1.1. Housekeeping reports

Every time the instrument is switched-on, and if the Housekeeping Report generation is not Disabled (via Service #3, sub type # 2), the instrument will generate a housekeeping report on a regular basis (as defined in the mission table, once every 2 to 60 seconds, nominal ( seconds). Total size of a Housekeeping Report packet is: 28 bytes.

The first HK report is sent one minute (60 seconds) after switch-on.

#### 6.3.1.2. Science reports (Only RTU-link)

While the instrument is in measurement (sounding mode), and if the Science Report generation (RTU-link) is not Disabled (via Service #20, sub type # 2), the instrument will generate a Science Report on a regular basis (as defined in the mission table, once every 2 to 60 seconds). Total size of a Science Report packet is: 1048 bytes

#### 6.3.1.3. Event reporting

An event packet is generated at achievement of each major step in the mode transitions:

- Hardware init performed
- Instrument Tuned
- Sounding mode started
- Sounding mode finished

#### 6.3.2. TM packet type summary

Process Id	Packet cat	APID dec	APID hexa	Packet Service Type dec	Packet Length (total)	Packet Type	Usage
59	1	945	3B1	1	20 (OK) 28 (Not OK)	ТМ	Telecommand acknowledge
59	4	948	3B4	3	28	ТМ	Housekeeping reports from Consert
59	7	951	3B7	5	24	TM	Event reporting from Consert
59	7	951	3B7	6	26	ТМ	Memory Check from Consert
59	7	951	3B7	17	16	ТМ	Test ping response from Consert
59	9	953	3B9	6	variable	ТМ	Memory Dump from Consert
59	12	956	3BC	20	1048	ТМ	Science Reporting from Consert

# 6.4. Telemetry Parameter definition

There are three different times for CONSERT:

- Rebuilt Time on ground : SCET Time (in SFDU Header)
- On-Board Set Time : OBT time
- CONSERT own Time: counter in TIC sets to zero when Consert is turned on and resets to zero after tuning phase, allows the precise synchronization between CONSERT Orbiter and CON. Lander

All TMs were dated with OBT time (standard TM format). HK and SCI TMs were dated with TIC.

#### 6.4.1. Successful Acknowledge

<b>Telemetry Packet I</b>	YCNST001				
Packet Name	CON_ACC_AC	K_SUCCESS	Instrument : Consert		
Packet Function	Consert Accepta	Consert Acceptance Acknowledge Success			
Generation Rules	s After reception of a valid TC Packet (with Ack field = 1)				
<b>Header Informatio</b>	n				
Process ID	59	Packet Category	1		
Service Type	1	Service Subtype	1		
Structure ID		Packet Length	Datafield = $10 + 4 = 14$		
		in Bytes	Total packet = 20		
<b>Data Field Informa</b>	tion				
Data Field	Field Structure	Remark			
NCNAST01	2 Bytes	MSB, lsb			
TC packet ID	-				
NCNAST02	2 Bytes	MSB , lsb			
TC Seq. control	-				
Notes :					

### 6.4.2. Failure Acknowledge

<b>Telemetry Packet I</b>	YCNST002 to YCNST007			
Packet Name	CON_ACC_FA	ILURE	Instrument : Consert	
Packet Function	Consert Accepta	ance Failure Report		
Generation Rules	After reception of	of an invalid TC Packet		
Header Information				
Process ID	59	Packet Category	1	
Service Type	1	Service Subtype	2	
Structure ID		Packet Length	Datafield = $10 + 12 = 22$	
		in Bytes	Total packet = 28	
Data Field Information				
Data Field	Field Structure	Remark		
NCNAST01	2 Bytes	MSB, lsb		
TC packet ID				
NCNAST02	2 Bytes	MSB , lsb		
TC Seq. control				
NCNAST03	2 Bytes	MSB, lsb, see note 1		
Failure Code				
Parameters	6 Bytes	See note 1		
Notes :				

#### Note 1 : failure code and parameter values

Failure	Failure name	Failure Reason	Param. 1	Param. 2	Par. 3	Par 4
Code			Byte	Byte	2 Bytes	2 Byte
1	ERR_TC_TIMEOUT	TC packet not	TC packet	TC	Nbr of expect	Nb of
		complete after 2	Туре	packet	Bytes (from	Bytes in
		seconds		SubType	TC Hd)	2 s
2	ERR_TYPE_WRONG CRC	Calculated CRC	TC packet	TC	CRC as read	CRC as
		is not egal to	Туре	packet	from packet	calculate
		CRC at end of		SubType	datafield	d using
		TC packet				TC data
3	ERR_TYPE_WRONGAPID	TC packet has	TC packet	TC	0	0
		wrong APID (ID	Туре	packet		
		# 59 or Cat #12)		SubType		
4	ERR_TC_TYPE_UNKNOWN	TC packet has	TC packet	TC	0	0
		unknown Type	Туре	packet		
		or Subtype		SubType		
5	ERR_TWO_MISS_TAB	TC with mission	TC packet	TC	0	0
		table received	Туре	packet		
		and other table		SubType		
		already received				
6	ERR_TC_DIRECT_UNKNOWN	Direct TC of	TC packet	TC	Direct TC	0
		unknown type	Туре	packet	value	
		received		SubType		

# 6.4.3. House-Keeping Report

Telemetry Packet InformationYCN00325			YCN00325	
Packet Name	CON_HK_REP		Instrument : Consert	
Packet Function	Consert Houseke	Consert Housekeeping parameter Report		
Generation Rules	Generated every	2 to 10 sec in all modes	k.	
<b>Header Informatio</b>	Header Information			
Process ID	59	Packet Category	4	
Service Type	3	Service Subtype	25	
Structure ID		Packet Length	Datafield = $10 + 12 = 22$	
		in Bytes	Total packet = 28	
Data Field Information				
Data Field	Field Structure	Remark		
Pad field	1 Byte	= 0		
SID	1 Byte	= 1		
NCNA0320	4 Bytes	HK_TIC B3 to B0: packet TIC (MSB to lsb))		
NCND033x	1 Byte	HK_STATUS : Instrument Status Byte		
NCND0339	1 Byte	HK_TEMP_OCXO : OCXO Temperature		
NCND0341	1 Byte	HK_TEMP_DIGI : digital board temperature		
NCND0342	1 Byte	HK_ADC_NBL : NBL level acquisition		
NCND0351	1 Byte	HK_ADC_TMIX : TMIX level acquisition		
NCND0352	1 Byte	HK_OCXO_SETTING : OCXO frequency setting		
Notes :				

#### Instrument status byte definition Value of global vaiable - status

Bit number Msbit = 7	information	Database Name	Name
7	0 = Init not performed 1 = init OK	NCND0331	STAT_BIT_INIT_OK
6	0 = Mission table not received 1 = Mission table received	NCND0332	STAT_BIT_MISS_TAB_OK
5	0 = Tuning not performed NCND0333 1 = Tuning performed		STAT_BIT_TUNING_OK
4	0 = Not in sounding mode 1 = In sounding mode NCND0334		STAT_BIT_SOUNDING
3	0 = Sounding not finished yet 1 = Sounding finished	NCND0335	STAT_BIT_END
2	0 = no HK reporting 1 = HK reporting enabled (default)	NCND0336	STAT_BIT_HKREP
1	0 = no SCreporting 1 = SC reporting enabled (default)	NCND0337	STAT_BIT_SCREP
0	0 = LOBT updated not received yet 1 = LOBT update received	NCND0338	STAT_BIT_LOBT

CONSERT	Refere Issue	nce: RO-OCN-TN-3044 : 3.2
Orbiter	Date Page	: 09/11/04 : 71 / 102
Experiment User Manual	Author	: J.P. Goutail

# 6.4.4. Progress Report

Telemetry Packet Information			YCN0A501 to YCN0D501		
Packet Name	CON_PROGRESS_REP		Instrument : Consert		
Packet Function	Consert Normal Progress Event report				
Generation Rules	After completion of important steps				
<b>Header Informatio</b>	Header Information				
Process ID	59	Packet Category	7		
Service Type	5	Service Subtype	1		
Structure ID		Packet Length	Datafield = 10 + 8 = 18		
		in Bytes	Total packet = 24		
Data Field Information					
Data Field	Field Structure	Remark			
NCNA0	2 Bytes	EID number from 41001 to 41004 (see note 2)			
EID					
NCND0511	1 Byte	OCXO_freq at end of tuning phase			
NCND0512	1 Byte	tuning_inter : confidence indicator of tuning phase			
		or 1 : good confidence			
		grater : bad S:N ratio			
NCND0513	1 Byte	Tuning phase GCW			
NCND0514	1 Byte	level GCW : ADC level achieved on NBL signal at			
		end of tuning phase AGC			
NCND0515	1 Byte	level_zero : ADC level achieved on NBL signal at end			
		of tuning phase, zero detection			
NCND0512	1 Byte	0 (Pad field)			
Notes :					

EID number	EID Name	Generation rule
41001	INITIALIZED	after completion of hardware init
41002	TUNING_OK	after completion of Tuning phase
41003	SOUNDING_STARTED	at start of sounding phase
41004	SOUNDING_COMPLETED	at end of sounding phase
	_	(instrument shall be switched off)

CONCEDT	Reference: RO-OCN-TN-3044		
CONSERI	Issue	: 3.2	
	Date	: 09/11/04	
Orbiter	Page	: 72 / 102	
Experiment User Manual	Author : J.F	P. Goutail	

# 6.4.5. Anomalous Report

Telemetry Packet Information YCN00502				
Packet Name	CON_ANO_EVENT		Instrument : Consert	
Packet Function	Consert anomal	Consert anomalous event report (Warning)		
Generation Rules	After detection of	of an anomalous Event		
Header Informatio	n			
Process ID	59	Packet Category	7	
Service Type	5	Service Subtype	2	
Structure ID		Packet Length	Datafield = $10 + 8 = 18$	
		in Bytes	Total packet = 24	
Data Field Information				
Data Field	Field Structure	Remark		
NCNA0EID	2 Bytes	EID number from 41007 to 41020 (see note 2)		
NCND0511	1 Byte	OCXO_freq at end of tuning phase		
NCND0512	1 Byte	tuning_inter : confidence indicator of tuning phase		
		or 1 : good confidence		
		grater : bad S:N ratio		
NCND0513	1 Byte	Tuning phase GCW		
NCND0514	1 Byte	level GCW : ADC level achieved on NBL signal at		
		end of tuning phase AGC		
NCND0515	1 Byte	level_zero : ADC level achieved on NBL signal at end		
		of tuning phase, zero d	letection	
NCND0512	1 Byte	0 (Pad field)		
Notes :				

#### Note 3 :

This events are only anomaly reports, no action has to be taken by S/C or ground on reception of these reports

<b>EID number</b>	EID Name	Generation rule
41007	TIMEOUT_AGC	FPGA reset due to timeout during AGC phase
41008	TIMEOUT_DATA	FPGA reset due to timeout during data transfer phase
41020	TUNING PB	Tuning phase algorithm has not converged
CONCEDE	Reference	: RO-OCN-TN-3044
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CONSERI	Issue	: 3.2
	Date	: 09/11/04
Orbiter	Page	: 73 / 102
Experiment User Manual	Author : J	.P. Goutail

### 6.4.6. Ping Test Report

<b>Telemetry Packet I</b>	Felemetry Packet InformationYCN01702			
Packet Name	CON_TEST_RE	SP	Instrument : Consert	
Packet Function	Consert test rep	ort (ping test)		
Generation Rules	After reception of	of a TC test request		
Header Informatio	n			
Process ID	59	Packet Category	7	
Service Type	17	Service Subtype	2	
Structure ID		Packet Length	Datafield = 10 + 0 = 10	
		in Bytes	Total packet = 16	
<b>Data Field Informa</b>	ition			
Data Field	Field Structure	Remark		
None				
Notes :				

CONCEDE	Reference	: RO-OCN-TN-3044
CONSERI	Issue	: 3.2
	Date	: 09/11/04
Orbiter	Page	: 74 / 102
Experiment User Manual	Author : J	I.P. Goutail

### 6.4.7. Science Report

Telemetry Packet Information YCN02003					
Packet Name	CON_SCI_REP		Instrument : Consert		
Packet Function	Consert Science	Report			
Generation Rules	Generated after e	each Sounding (every 2	to 10 s) in Sounding Mode		
<b>Header Informatio</b>	n				
Process ID	59	Packet Category	12		
Service Type	20	Service Subtype	3		
Structure ID		Packet Length	Datafield = $10 + 1032 = 1042$		
		in Bytes	Total packet = 1048		
<b>Data Field Informa</b>	tion				
Data Field	Field Structure	Remark			
Parameter 1	4 Bytes	SC_TIC B3 to B0: so	ounding start TIC MSB to lsb		
Parameter 2	1 Byte	SC_TEMP_OCXO : O	CXO Temperature		
Parameter 3	1 Byte	SC_TEMP_DIGI : dig	ital board temperature		
Parameter 4	2 Byte	SC_SOUNDING_N E	1; B0: sounding number MSB to		
		lsb			
Parameter 5	1 Byte	SC_GCW : Gain control	ol word of this sounding		
Parameter 6	1 Byte	SC_OCXO_SETTING	: OCXO frequency setting Byte		
Parameter 7	510 Bytes	SC_SIGNAL_I (	(MSB ; lsb) for position 0 to 254		
Parameter 8	510 Bytes	SC_SIGNAL_Q (	MSB ; lsb) for position 0 to 254		
Parameter 9	2 Bytes	Spare bytes			
Notes :	Notes :				

### 6.4.8. Memory Dump Report

Telemetry Packet InformationYCNC00606				
Packet Name	CON_MEMO_I	DUMP	Instrument : Consert	
Packet Function	Consert Memor	y dump Telematry		
Generation Rules	After reception of	of a telemetry dump req	uest TC	
Header Information	n			
Process ID	59	Packet Category	9 (Dump)	
Service Type	6	Service Subtype	6	
Structure ID		Packet Length	Datafield = 10 + 8 + length *2	
		in Bytes	Total packet = $16 + 8 + \text{length}$	
Data Field Informa	tion		· 2	
Data Field	Field Structure	Remark		
NCND0600	1 Byte	Always 60 (dec) for Consert Memory		
Memory_ID				
NCND0610	1 Byte	Always 1 for Consert		
N number of blocks				
NCNA0610	4 Bytes	MSB to lsb, the two fi	irst bytes are always 00	
Start Address		(only64k octet mem)		
NCNA0620	2 Bytes	MSB lsb, length in 16	bit words $=$ n	
Block length = $n$				
NCNA0630	2n Bytes	Dumped memory		
Data				
Notes : Only one valid Memory ID for Consert = 60 dec				
Only one memory segment can be dumped at each time				

### 6.4.9. Memory Check Report

Telemetry Packet Information YCNC00610				
Packet Name	CON_MEMO_N	MEM_CHECK_TM	Instrument : Consert	
Packet Function	Consert Memory	y dump Telemetry		
Generation Rules	After reception of	of a memory check calle	culation request TC	
<b>Header Information</b>	n			
Process ID	59	Packet Category	7 (event)	
Service Type	6	Service Subtype	10	
Structure ID		Packet Length	Datafield = 10 + 10	
		in Bytes	Total packet = $16 + 10$	
<b>Data Field Informa</b>	tion			
Data Field	Field Structure	Remark		
Memory_ID	1 Byte	Always 60 (dec) for Consert Memory		
N number of blocks	1 Byte	<b>Always 1 for Consert</b>		
Start Address	4 Bytes	MSB to lsb, the two fi	rst bytes are always 00	
		(only64k octet mem)		
Block length $=$ n	2 Bytes	MSB lsb, length in 16	bit words = $n$	
Data2 BytesCRC16 value of the designated memory area				
Notes : Only one valid Memory ID for Consert = 60 dec				
Only one memory segment can be dumped at each time				
· · ·	-	-		

Experiment User Manual

Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 77 / 102 Author : J.P. Goutail

### 6.5. Event Packet definition

Normal progress packet definition see part 6.4 for packet definition

### 6.6. Context file Definition

N/A not used by Consert

### 6.7. Data and Dump File Definition

see TM and TC packet definition (6.2 & 6.4)

#### Orbiter

Experiment User Manual

### Annex 1 : Consert Orbiter Instrument Bench Test Procedure

- 1. Switch-on the Consert E Box.
- 2. Wait 10 seconds
- 3. Send Time-update (button in Remote window)
- 4. Wait 25 seconds
- 5. Perform ping test (button in Remote window)
- 6. Check Ping success on SIS Log window
- 7. Check Ping Event report on lecture\_dat
- 8. Wait for first HK packet , check values in Lecture\_dat, (default HK distribution, if no Mission table send before, is 60 seconds, with new HK packet every 15 seconds)
- 9. Send a mission table :
  - Select a TC directory in the default EMC file

Type a valid TC name : MTAFT (for Mission table for AFT) with following settings :

TuneTIC (ie time between power on and tuning start) = 180 sec.

Tuning duration will be : around 20 seconds

StartTIC (ie time between end of tuning and sounding start) = 60 sec

DeltaTIC (time between soundings) = 5 seconds

Nbsoundings = 100 total numer of soundings

Total duration until sounding stop = 180 + 20 + 60 + 5 \* 100 = 12 minutes

Orbiter

Experiment User Manual

#### Annex 2 : Consert I&T – UFT procedure for EQM

Note : This Procedure was not performed on the Flight Model.

10. Start-up 20. Main Power 20.10. Select Main Data I/F 20.20. Send ZPWMA111SWON (Switch-on Consert) 20.30. Check Parameters : LCL Voltage LCL Current (Initialization performed) NCND0331 NCND0339 (OCXO Temperature) (DIGI Board Temperature) NCND0341 20.40. Wait 40 sec 20.50. Send SCET update 20.60. Check 20.70. Wait NCND0338 = 1 (LOBT Received) 20.80. Send Private TC : ZCN19201 with PCNGA010 = 0x0100(Mission Table index) (Start TIC for Tuning Phase) PCNGA020 = 0x0001AD27(Start TIC for Soundings) PCNGA030 = 0x00008F0DPCNGA040 = 0x0BEC(Time Step in TICs) (Total Number of Soundings) PCNGA050 = 0x0064(Clock initial Frequency + Mode Setting) PCNGA060 = 0x8000PCNGA070 = 0x001F(GCW Attenuations) PCNGA080 = 0x9585(TAB NBL Settings) 20.90. Wait NCNA0520 = 1 NCNA0520 (Sounding Completed) During the execution of the mission table monitor the following parameters on CCS console: NCND0332 (Mission Table Received) NCND0333 (Tuning Phase performed) NCND0334 (Sounding Continuing) NCND0339 (OCXO Temperature) NCND0341 (DIGI Board Temperature) NCND0351 (TMIX Level Acquisition) NCNA0320 (Packet TIC Value) 20.100. Send ZPWMA112SWOF (Switch-off Consert) 30. Redundant Power 30.10. Select Redundant Data I/F 30.20. Send ZPWMA368SWON (Switch-on Consert) 30.30. Check the same Parameters as for step 20.30 30.40. Wait 30 sec 30.50. Send SCET update 30.60. Check 30.70. Wait NCND0338 = 1 30.80. Send Ping: ZCN01701 30.90. Send ZPWMA369SWOF (Switch-off Consert) 40. Final activities

Orbiter

Experiment User Manual

### Annex 3 : Consert IST procedure for PFM

Test	Step	Operations
0		TEST CONFIGURATION CHECK-UP
0	10	CCS check: Verify that Open Center is up, and the test environment is con-nected to an active session. Synoptics to monitor S/C status are active.
0	20	CONSERT EGSE check: Verify that CONSERT EGSE is ON, the test software is started and ready to accept the connection from the CCS.
0	30	Establish CCS - CONSERT EGSE connection and from CONSERT EGSE, request the Telemetry Delivery of the following APIDs: 945, 948, 951, 953, 956, 1804
0	40	At the Sequence Monitor launch the procedure: Exec POWER ON
0	50	At the first prompt asking to select power configuration select: BATTERY Simulation
0	60	When prompted whether the PDU-P/L has to be switched on reply: "YES"
0	70	When prompted whether the RTU-P/L has to be switched on reply: "YES"
0	80	Spacecraft configuration: Verify that the S/C is powered and the following units are also powered and delivering telemetry to ground: DMS PDU-SS PDU-PL RTU-SS RTU-PL SSMM
0	90	Start the CONSERT IST procedure for EQM launching the sequence: Exec PCNISTF0
0	100	Answer to the question asking to verify the correct configuration. "ARE SATELLITE AND GSE CONIFIGURED AS DES-CRIBED? YES/ <no>"</no>
0	110	Answer to the question: "IS THE RF OUTPUT CONNECTED TO THE ANTENNA? YES/ <no> [DUMMY LOAD = NO]"</no>
0	120	Answer to the question: "DID YOU ASK FOR TELEMETRY REPORT? YES/ <no>"</no>
0	130	On the log window messages are shown confirming that: the Nominal Interface has been successfully selected

### Orbiter

Test	Step	Operations
0	140	On the log window messages are shown confirming that: TC delivering has been successfully enabled
0	150	On the log window messages are shown confirming that: TM polling has been successfully enabled
1		POWER ON
1	10	Once the master sequence gives the control to the child sequence PCNISTF0SPON, the description of the activities is presented on the screen and the users is requested to confirm the test execu-tion. Answer to the question: "DO YOU WANT TO PROCEED AND EXECUTE THIS STEP? <yes>/NO"</yes>
1	20	Answer to the question: "DO YOU WANT TO USE <m>AIN OR <r>EDUNDANT POWER INTERFACE?" selecting the main power interface (typing: M)</r></m>
1	30	On the Log window messages are displayed confirming that the power on Consert (nominal) and the LCL status are cheked.
1	40	On the log window a message is shown confirming that: CONSERT H/K packets are correctly received.
1	50	On the log window the message "CONSERT Boot event received !!!" confirming the reception of the boot event is displayed and an acoustic signal is generated by the CCS.
1	60	On the log window a message is shown confirming that: the SCET update service is started and the related TC has been accepted.
1	70	On the log window a message is shown confirming that: CONSERT has reached the "Wait Mission Table" state commanded in scientific mode via the TC 'Reset Task' (230,3) and the related TC has been accepted.
1	80	On the log window is displayed the actual power consumption: Verify that the values of the parameters displayed are within the nominal ranges/values and answer to the question: "Are the above displayed parameter values correct? YES/NO"
1	90	An acoustic signal generated by the CCS evidences that CONSERT has been powered and the execution control has been given back to the master sequence.
2		CONNECTION TEST
2	10	Once the master sequence gives the control to the child sequence PCNISTF0S017, the description of the activities is presented on the screen and the users is requested to confirm the test execu-tion. Answer to the question: "DO YOU WANT TO PROCEED AND EXECUTE THIS STEP? <yes>/NO"</yes>
2	20	On the log window a message is shown confirming that: the connection test has been requested (17,3).

### Orbiter

Test	Step	Operations
2	30	On the log window a message is shown confirming that the connection test response has been correctly generated: Verify that it has been received by answering to the question: "Is the above displayed Connection Test Report Response correct? YES/NO"
2	40	Verify that the connection test response has successfully received at CONSERT EGSE as well.
2	50	An acoustic signal generated by the CCS evidences that the execution control has been given back to the master sequence
3		TM/TC VERIFICATION
3	10	N/A All the usable not–private have been already used in the other tests of this IST
4		PATCH AND DUMP
4	10	Once the master sequence gives the control to the child sequence PCNISTF0SPTC, the description of the activities is presented on the screen and the users is requested to confirm the test execution. Answer to the question: "DO YOU WANT TO PROCEED AND EXECUTE THIS STEP? <yes>/NO"</yes>
4	20	Verify that CONSERT still generates H/K packets.
	30	Verify that the housekeeping telemetry is disabled
4	40	MemID 60 "RAM" (1 Block) Verify that the patch memory TC (6,2) has been sent and no errors are displayed in the Log SequenceWindow.
4	50	MemID 60 "RAM" (1 Block) Verify that the dump memory report (6,6) has been received and checked and no errors are displayed in the Log SequenceWindow.
4	60	MemID 60 "RAM" (1 Block) Verify that the check memory report (6,10) has been received and its contents are compared with expected ones. Verify that no errors are displayed in the Log SequenceWindow.
4	70	MemID 60 "RAM" (2 Blocks) Verify that the patch memory TC (6,2) has been sent and no errors are displayed in the Log SequenceWindow.
4	80	MemID 60 "RAM" (2 Blocks) Verify that the dump memory report (6,6) has been received and checked. Verify that no errors are displayed in the Log SequenceWindow.

### Orbiter

Test	Step	Operations
4	90	MemID 60 "RAM" (2 Blocks) Verify that the check memory report (6,10) has been received and its contents are compared with expected ones. Verify that no errors are displayed in the Log SequenceWindow.
4	100	MemID 60 "RAM" (3 Blocks) Verify that the patch memory TC (6,2) has been sent and no errors are displayed in the Log SequenceWindow.
4	110	MemID 60 "RAM" (3 Blocks) Verify that the dump memory report (6,6) has been received and checked and no errors are displayed in the Log SequenceWindow.
4	120	MemID 60 "RAM" (3 Blocks) Verify that the check memory report (6,10) has been received and its contents are compared with expected ones and no errors are displayed in the Log SequenceWindow.
4	130	In case of errors, at the question "THE TEST WAS NOT OK. DO YOU WANT TO REPEAT IT? <yes>/NO"</yes>
4	140	Verify that CONSERT still generates H/K packets
4	150	An acoustic signal generated by the CCS evidences that the execution control has been given back to the master sequence
5		DMS-P/L TRAFFIC MANAGEMENT
5	10	Once the master sequence gives the control to the child sequence PCNISTEOSDMS, the description of the activities is presented on the screen and the users is requested to confirm the test execution. Answer to t the question: "DO YOU WANT TO PROCEED AND EXECUTE THIS STEP? <yes>/NO"</yes>
5	20	On the log window a message is shown confirming that: Disable H/K report (3,6) has been sent and the related TC has been accepted.
5	30	On the log window a message is shown confirming that: Reset TM Output Buffer (255,1) has been sent and the related TC has been accepted.
5	40	After approximately 2 minutes verify that On the log window a message is shown stating that <b>NO</b> CONSERT H/K packets have been received.
5	50	Verify that CONSERT EGSE did not receive H/K packets as well and answer at the question "Do you confirm that no H/K packets have been collected by CONSERT SCOE? YES/NO"
5	60	On the log window a message is shown confirming that: Enable H/K report (3,5) has been sent and the related TC has been accepted.
5	70	After approximately 2 minutes verify that On the log window a message is shown stating that <b>SEVERAL</b> CONSERT H/K packets have been received.

### Orbiter

Test	Step	Operations
5	80	Verify that CONSERT EGSE received H/K packets as well and answer at the question: "Do you confirm that several H/K packets have been collected by CONSERT SCOE? YES/NO"
5	90	An acoustic signal generated by the CCS evidences that the execution control has been given back to the master sequence
6		CONTEXT (Not supported by DMS SW 1.2.2)
	10	
7		SCIENCE DATA TRANFER
7	10	Once the master sequence gives the control to the child sequence PCNISTEOSCIE, the description of the activities is presented on the screen and the users is requested to confirm the test execution. Answer to t the question: "DO YOU WANT TO PROCEED AND EXECUTE THIS STEP? <yes>/NO"</yes>
7	20	On the log window a message is shown confirming that: Enable Science via RTU (20,1) has been sent and the related TC has been accepted.
7	30	On the log window a message is shown with the actual LCLs 52A and 52B Status and current consumption
7	40	On the log window a message is shown confirming that: "Start Mission Table" TC (192,1) has been sent and the related TC has been accepted.
7	50	On the log window a message is shown confirming that: CONSERT has received the mission table and it has reached the "Mission Table received" state.
7	60	On the log window a message is shown confirming that: CONSERT has finished the tuning phase and has reached the "Tuning Phase Performed" state.
7	70	On the log window a message is shown with the actual LCLs 52A and 52B Status and current consumption
7	80	On the log window a message is shown confirming that: CONSERT reached the "Sounding Continuing" state.
7	90	Several connection test are sent to CONSERT in order to exercise the TC channel as well.
7	100	On CONSERT EGSE, check that Science TM is correctly received and displayed by answering to the question: "Is Scientific telemetry correctly received by CONSERT EGSE? <yes>/NO"</yes>
7	110	On the log window a message is shown with the actual values of OCXO Temperature, DIGI board Temperature and TMIX level Acquisition
7	120	On the log window a message is shown with the actual LCLs 52A and 52B Status and current consumption
7	130	On the log window a message is shown informing that OCXO frequency has been changed and set to 170 (0xAA).
7	140	On the log window a message is shown informing that OCXO frequency has been changed back to its original values of 128 (0x80).

### Orbiter

Test	Step	Operations
7	150	On the log window messages are displayed informing of different packets receiving.
7	160	On the log window a message is shown confirming that: the event 'Sounding Completed' has been correctly generated and successfully received.
7	170	On the log window a message is shown confirming that: Disable Science via RTU (20,2) has been sent and the related TC has been accepted.
7	180	On the log window a message is shown with the actual LCLs 52A and 52B Status and current consumption
7	190	An acoustic signal generated by the CCS evidences that the execution control has been given back to the master sequence
8		PRIVATE SERVICES
8	10	N/A All private TC are used in the PCNISTE0SCIE routine
9		OPERATIVE MODES
9	10	N/A All possible operative modes are used all over the IST
10		MAX DATA THROUGHPUT
10	10	N/A The max data throughput is reached during science data production (see test 7)
11		POWER CONSUPTION
11	10	N/A The max power consumption is monitored within the test 7 (Science data transfer)
12		FUNCTIONAL PERFORMANCE TEST
12	10	N/A The Functional Performance Test is performed within the tests 8 (Science data transfer)
13		POWER OFF
13	10	Once the master sequence gives the control to the child sequence PCNISTF0SOFF, the description of the activities is presented on the screen and the users is requested to confirm the test execution. Answer to the question: "DO YOU WANT TO PROCEED AND EXECUTE THIS STEP? <yes>/NO"</yes>
13	20	On the log window a message is shown confirming that: It is checked the CONSERT state before Power Off.
13	30	Verify that the values of the parameters displayed in the log window are within the nominal ranges/values and, if correct, type <yes> to Power Off CONSERT.</yes>
13	40	On the log window is displayed the message confirming that SCET update (stop) has been sent.
13	50	On the log window the message confirming that both CONSERT LCL's are OFF is displayed.
13	60	An acoustic signal generated by the CCS evidences that the execution control has been given back to the master sequence
14		REDUNDANCY

### Orbiter

Test	Step	Operations
14	10	On the log window messages are shown confirming that: the Redundant Interface has been successfully selected
14	20	On the log window messages are shown confirming that: TM polling has been successfully enabled
14	30	On the log window messages are shown confirming that: TC delivering has been successfully enabled
14	40	Once the master sequence gives the control to the child sequence PCNISTF0SPON, the description of the activities is presented on the screen and the users is requested to confirm the test execution. Answer to the question: "DO YOU WANT TO PROCEED AND EXECUTE THIS STEP? <yes>/NO"</yes>
14	50	At the question "DO YOU WANT TO USE <m>AIN OR <r>EDUNDANT POWER INTERFACE?" select the redundant power interface by typing:</r></m>
14	100	Verify that the Consert Power (redundantl) and LCL status are setted to: On while for nominal lines: Off
14	60	On the log window a message is shown confirming that: H/K packets are correctly generated by CONSERT.
14	70	On the log window the message "CONSERT Boot Successful!!!" confirming the reception of the boot event is displayed and an acoustic signal is generated by the CCS.
14	80	On the log window a message is shown confirming that: the SCET update service is started and the related TC has been accepted.
14	90	Verify that CONSERT EGSE receives and H/K packets as well and their contents is correctly displayed
14	100	Verify that the values of the parameters displayed in the log window are within the nominal ranges/values and confirm it by typing.
14	110	An acoustic signal generated by the CCS evidences that CONSERT has been successfully powered and the execution control has been given back to the master sequence.
14	120	Once the master sequence gives the control to the child sequence PCNISTF0S017, the description of the activities is presented on the screen and the users is requested to confirm the test execution. Answer to the question: "DO YOU WANT TO PROCEED AND EXECUTE THIS STEP? <yes>/NO"</yes>
14	130	On the log window a message is shown confirming that: the connection test has been requested (17,3) and the related TC has been accepted.
14	140	On the log window a message is shown confirming that: the connection test response has been correctly generated and successfully received.
14	150	An acoustic signal generated by the CCS evidences that the execution control has been given back to the master sequence

### Orbiter

Test	Step	Operations
14	160	Once the master sequence gives the control to the child sequence PCNISTF0SOFF, the description of the activities is presented on the screen and the users is requested to confirm the test execution. Answer to the question: "DO YOU WANT TO PROCEED AND EXECUTE THIS STEP? <yes>/NO"</yes>
14	170	Answer to the question: "Check the above displayed parameter and type <'YES'> to power OFF CONSERT, 'NO' to ABORT" check the values of the requested parameters and if correct type:
14	180	On the log window the message confirming that both CONSERT LCL's are OFF is displayed.
14	190	An acoustic signal generated by the CCS evidences that the execution control has been given back to the master sequence
15		SHUT DOWN
15	10	On the log window messages are shown confirming that TM polling has been successfully disabled
15	20	On the log window messages are shown confirming that: TC delivering has been successfully disabled
15	30	Follow the execution of the sequence until the end; When prompted, to disconnect the POWER and TMTC SCOEs answer : N
15	40	From CCS console, switch OFF the P/L-RTU by sending the following device commands: {ZDMC0053SWOF} and {ZDMC0055SWOF}
15	50	Switch OFF the S/C-RTU by executing at CCS console the sequence "EXEC POWER_RTU"
15	60	Switch OFF the Satellite from the POWER SCOE console

Orbiter

Experiment User Manual

### Annex 4 : Consert Orbiter Instrument Full Functional Test Procedure

	Test Date and	<u>Time :</u>									
1	Local Date and Time										
2	SIS Date and Time										
3	EGSE Date and Time										
	r 1511										
4	Log File name										
	Initialisation & First	<u>Measurements</u>									
5	Switch Consert on Nominal Power										
6	Read out primary power										
	(circa 200 mA going down to 93 mA ir	n 10 seconds)									
	Check on "lecture_dat_v5.vi" window :										
	"Init OK", (Green Button)										
7	On the remote window, click on "Time Upload"										
-	(Starts time synchronization service 9 for ID 59)										
8	Verify message on Log Window :	• \									
•	"Time Packet Transmit Confirmed", (Green Li	ine)									
9	Uneck on "lecture_dat_v5.v1" window : "LOPT Lindata ON" (Green Putten)										
10	On the "remote" window click on " <b>Ping Test</b>	11									
10	(Starts test TC service 17 for ID 59)										
11	Check on "lecture dat v5 vi" window :										
	"EID name : Ping Test OK"										
12	On the remote window, Select TCFILEname	= FFT									
13	Early HK packet : PCK Number										
	Status bit setting Init	Green									
	НК	Green									
	SC	Green									
	LOBT	Green									
	Mission Table	Red									
	Tuned	Red									
	Sounding	Red									
	Finished	Red									
	TIC	1 min 0 sec									
	OCXO setting	128									
	Digi temp	173									
	OCXO Temp	171									

### Orbiter

	HouseKeeping Reporting Disabling	
14	Type "STOPHK" in "TCname" field and hit "TC request"	
	(Disables HK distribution)	
15	Verify TC acceptance message on SIS :	
	"DMS Service 1 Acceptance ACK"	
16	Wait 30 seconds and verify that no HK packets are generated	
17	Type "STARTHK" in "TCname" field and hit "TC request"	
	(Enables HK distribution)	
18	Verify TC acceptance message on SIS :	
	"DMS Service 1 Acceptance ACK"	
19	Verify that HK packets are generated again ("HK packet num"	
	increasing)	
	Wrong Commands not acconted	
	wrong Commands not accepted	
20	Type "WRAPID" in "TCname" field and hit "TC request"	
	(Wrong APID Number)	
21	Verify error message on SIS :	
	"DMS Service 1 Acceptance NAK : Incorrect APID (Code 3)", (Red	
	Line)	
22	Type "WRTYPE" in "TCname" field and hit "TC request"	
	(Wrong Type)	
23	Verify error message on SIS :	
	"DMS Service 1 Acceptance NAK : Invalid Command (Code 4)",	
	(Red Line)	
	Mission Table seconds describe second	
	wission Table accepted only once	
24	Type "MTAB1" in "TCname" field and hit "TC request"	
	(Sends Mission Table 1)	
25	Verify TC acceptance message on SIS :	
	"DMS Service 1 Acceptance ACK"	
26	Check on lecture dat v5.vi Window :	
	"Mission Table ON", (Green Button)	
27	Type "MTAB1" in "TCname" field and hit "TC request"	
	(Sends again a mission table)	
28	Verify error message on SIS :	
	"DMS Service 1 Acceptance NAK : Cannot execute Command (Code	
	5)", (Red Line)	

Orbiter

Experiment User Manual

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	Tuning Failed Message										
	Note : This part of the test MUST be done du	uring Tuning Mode of the	instrument.								
29	Wait for instrument to be in tuning mode										
	(About 2.5 minutes waiting time).										
	(Current increase on primary supply fro	om 100 mA to 220 mA.									
30	Log Primary Current during Tuning	220 mA									
31	On the SIS, Open "TM Block Monitor" in	"View Menu"									
	(Displays the frame content of TM)										
32	Hit "Single-Shot" button on TM Block Monit	tor window to stop									
	Display sampling mode										
33	On the EGSE,	( EID 410201									
	Verify distribution of Anomalous Progress Re	port, $EID = 41020d$									
24	"Tuning FAILED"										
34	Walt 30 seconds	FID = 410024									
32	Verify distribution of Normal Progress Report	, EID = 41003a									
26											
30	Log. Frequency after tuning	220									
	Intercartille	5 (if no Lander)									
	GCW	0 (if no Lander)									
	NBL after GCW	129 (if no Lander)									
	NBL at stop	129 (if no Lander)									
37	On the SIS.										
-	Hit "Continuous" button on TM Block Moni	tor window to go in RT									
	mode again	e									
38	Verify distribution of TM block with HK and	SC packets.									
	Science Reporting	<u>e Disabling</u>									
39	Type "STOPSC" in "TCname" field and hit "	TC request"									
	(Disable Science Packet Generation)										
40	Wait 30 seconds and verify that only HK pack	ets are generated									
41	Type "STARTSC" in "TCname" field and hit	t "TC request"									
4.5	(Enable Science Packet Generation)										
42	Verity that SC packets are generated again										

### Orbiter

Experiment User Manual

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	Miscellaneous	
43	Type "RESET" in "TCname" field and hit "TC request"	
	(Disable Science Packet Generation)	
44	Hit "Ping Test" within 5 seconds	
	(Starts test TC service 17 for ID 59)	
45	Verify Error message on SIS :	
	"Service 17 Test Failed : Test Timed out", (Red Line)	
46	Wait for SC distribution	
47	Hit "Stop APID report"	
	(Stops Polling)	
48	Wait 20 seconds	
49	Hit "Start APID report"	
50	(Starts Polling)	
50	Verify that no error is logged on SIS	
51	Switch Off Consert	
	<u>Redundant Mode</u>	
52	On the EGSE, Hit "Stop APID report"	
	(Stops Polling)	
53	On the EGSE, Hit "Stop Session"	
	(Stops Session)	
54	On the EGSE, Select "Redundant" in "Channel" field and hit "Start	
	Session"	
55	Verify message on Log Window :	
- (	"TC Routing Mode Set to REDUNDANT"	
56	Check on ROSIS System Log Window :	
	"HLBC Enabled Polling of Packet Terminal 0", (Green Line)	
57	Check on ROSIS System Log Window : "HLBC BCP3 Event	
50	OBI=****** (Green Line)	
30	On the remote window, click on "Start AriD report"	
59	Check on ROSIS System Log Window :	
	"TM-D Handler Message: Enabled APIDs: 948, 951, 956", (Green	
	Line)	
60	Switch-on Consert on Redundant Channel	
61	Read out primary power (circa 200 mA going down to 93 mA in 10 seconds)	
62	On the remote window, click on "Time Upload"	
	(Starts time synchronization service 9 for ID 59)	
63	Verify message on Log Window :	
	"Time Packet Transmit Confirmed", (Green Line)	
64	Check on "lecture_dat_v5.vi" window :"LOBT Update ON", (Green	

### CONSERT Orbiter

Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 92 / 102 Author : J.P. Goutail

	Button)		
65	On the "remote" window, click on "Ping Test	**	
	(Starts test TC service 17 for ID 59)		
66	Check on "lecture_dat_v5.vi" window :		
	"EID name : Ping Test OK"		
67	Early HK packet : PCK Number		
	Status bit setting Init	Green	
	HK	Green	
	SC		
	LOBT		
	Mission Table	Red	
	Tuned	Red	
	Sounding	Red	
	Finished	Red	
	TIC	1 min 0 sec	
	OCXO setting	128	
	Digi temp	173	
	OCXO Temp	171	
<b>68</b>	Switch Off Consert		
	End of Te	est	
69	Hit "Stop APID report"		
	(Stops Polling)		
70	Hit "Stop Session"		
	(Stops Session)		
71	Hit "Terminate Link"		
	(Stops Polling)		
72	Verify message on Log Window :		
	"System Off-Line Mode Established"		
73	On the SIS, Hit "Local Commanding Mode"		
74	Type Password : "ROSIS"		
75	Hit "Exit" in "File" Monu		
-	FIL EXIL III FILE MIEILU		
76	Switch-Off EGSE		

Experiment User Manual

Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 93 / 102 Author : J.P. Goutail

## Annex 5 : Consert Orbiter Instrument Data Description

						Pa	cket	Hea	der						
							Pack	et ID							
V	version	#	Type	Data				A	pplica	tion Pr	ocess II	D			
				Field	Process ID								Packet C	Categor	y
				Head											
				er											
	0		0	Flag									<u> </u>	[	1
0	0	0	0	I	0				0	I	l				
C	· .·					Packe	<u>t Sequ</u>	ence C	<u>ontrol</u>						
Segmentati     Source Sequence Count       on Flags															
1	1														
	Packet Length = (4096 + 10) - 1) Max														
	Packet Data Field														
	Data Field Header														
							SCET	Time							
						e e	Second	S							
							Second	S							
						Fracti	onal Se	conds							
Packe	et Utilis	sation	Chec		Sp	are				Pac	ket Sei	rvice T	ype		
S	Standar	d	ksum												
			Flag										1		
			De alvat (	Subtra				Dev	d Eiald	or	Dad fia	ldafa	almarvi	adaad	
		1		Subtyp				ra		- 0 01	r au ne			euged	
					Se	urce F	)ata (4	096 Rv	tes Ma	x)					
					50		( T		IVEG						

Orbiter

Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 94 / 102 Author : J.P. Goutail

#### Experiment User Manual

### ETM00101 / CON\_ACC\_ACK\_SUCCESS

	15	14	13	12	11	10	9	8	7	6	5	4	3	2	1	0
0	V	ersion	=0	0	1			Proc	ess ID	tegory	= 1					
1	1	1						Sourc	e Sequ	uence (	Count					
2	Packet Length = 13															
3	OBT Time (Seconds)															
4	OBT Time (seconds)															
5						OI	BT Tin	ne (fra	ctional	secon	ds)					
6		PUS		Chk		Sp	are				Pack	et Serv	ice typ	e = 1		
7		]	Packet	Servic	e subt	ype = 1	1					PAD	Field			
8								TC Pa	cket ID	)						
9							TC	Sequer	nce Con	ntrol						

### ETM00102 / CON\_ACK\_FAILURE

	15	14	13	12	11	10	9	8	7	6	5	4	3	2	1	0		
0	V	ersion	=0	0	1			Proc	cess ID = 59 Packet Category = 1									
1	1	1						Sourc	irce Sequence Count									
2							Pac	cket Le	ength =	= 21								
3	OBT Time (Seconds)																	
4	OBT Time (seconds)																	
5	OBT Time (fractional seconds)																	
6		PUS		Chk		Sp	are		Packet Service type = 1									
7		]	Packet	Servic	e subty	ype = 2	2				I	PAD F	ield =	0				
8							1	TC Pa	cket ID	)								
9							TC	Sequer	nce Co	ntrol								
10								Failur	e Code	;								
11				Param	eter 1				Parameter 2									
12				Param	eter $\overline{3}$				Parameter 4									
13				Param	eter 5				Parameter 6									

Orbiter

Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 95 / 102 Author : J.P. Goutail

#### Experiment User Manual

## ETM00325 / CON\_HK\_REP

	15	14	13	12	11	10	9	8	7	6	5	4	3	2	1	0
0	V	ersion=	=0	0	1			Proc	ess ID	= 59			Pac	ket Ca	tegory	= 4
1	1	1						Sourc	e Sequ	uence (	Count					
2							Pac	cket Le	ngth =	21						
3							OB	Г Time	(Seco	nds)						
4							OB	T Time	e (seco	nds)						
5				-		OI	3T Tin	ne (frac	ctional	secon	ds)					
6	PUS     Chk     Spare     Packet Service type = 3															
7	Packet Service subtype = 25 PAD Field = 0															
8	PAD Field = 0 Structure ID = 1															
9				0	n-Boa	rd Tim	ie in T	ICs MS	SW (6	5536 1	Tics =	107 se	c)			
10					On-Bo	oard Ti	me in	TICs L	LSW (*	7706 T	ics = 1	13 sec)				
11	Init	Miss	Tuni	Soun	Soun	ΗK	SC	LOB			OC	XO Te	mpera	ture		
	OK	ion	ng	ding	ding	Rep	Rep	Т								
		Tabl	OK	Start	Finis	Enab	Enab	Rece								
		e		ed	hed			ived								
		OK														
12		]	Digital	Board	Temp	erature	e				NBL	Level	Acqui	sition		
13			TMIX	K Level	Acqu	isition					OCXC	) Frequ	lency S	Setting	,	
م م	1 000	D 001				4002	1000	0001	0001	0504	<b>07 3</b> D		0 1 0 1			

0BB4 C00D 0015 0000 00D4 A000 4003 1900 0001 0001 C504 C7 AB AD 80 12 50 Len Time Type OBT St Temp

Orbiter

Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 96 / 102 Author : J.P. Goutail

#### Experiment User Manual

### $ETM00501 \ / \ CON\_PROGRESS\_REP$

	15	14	13	12	11	10	9	8	7	6	5	4	3	2	1	0	
0	V	ersion	=0	0	1			Proc	cess ID = 59 Packet Category = 7								
1	1	1						Sour	Irce Sequence Count								
2	Packet Length = 17																
3	OBT Time (Seconds)																
4	4 OBT Time (seconds)																
5	OBT Time (fractional seconds)																
6		PUS		Chk		Sp	are				Pack	et Serv	ice typ	ce type = $5$			
7		]	Packet	Servic	e subt	ype = 1	[		PAD Field								
8		Ever	nt ID (•	41002	= Tuni	ng OK	L, 4100	3 = Sc	ounding	g starte	ed, 410	04 = S	oundir	ng finis	shed)		
9			Cloc	k Frequ	uency	(or 0)					In	itercati	lle (or	0)			
10			Tu	ning pł	nase Go	CW			Level GCW								
11				Level	Zero						Р	AD Fi	eld (= 0)	0)			

### ETM00502 / CON\_ANO\_EVENT

	15	14	13	12	11	10	9	8	7	6	5	4	3	2	1	0
0	V	ersion	=0	0	1			Proc	ess ID	= 59			Pac	ket Ca	tegory	= 7
1	1	1						Sour	ce Sequ	uence (	Count					
2							Pa	cket Le	ength =	- 17						
3	OBT Time (Seconds)															
4	OBT Time (seconds)															
5	OBT Time (fractional seconds)															
6		PUS		Chk		Sp	are				Pack	et Serv	ice typ	e = 5		
7		]	Packet	Servic	e subt	ype = 2	2					PAD	Field			
8		]	Event 1	ID (410	007 = 7	Гітеоі	ut AG	C, 4100	)8 = Ti	meout	Data,	41020	= Tun	ing Pb	)	
9			Cloc	k Frequ	uency	(or 0)					In	tercati	lle (or	0)		
10			Tu	ning pł	nase G	CW						Level	GCW			
11				Level	Zero						Р	AD Fie	eld (= 0)	0)		

0BB7 C005 <u>0011</u> <u>0000 00D4 A000</u> 40<u>05 01</u>00 <u>A02B</u> DC08 0081 8100 Time Type EID# CONSERT Orbiter Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 97 / 102 Author : J.P. Goutail

Experiment User Manual

### ETM00610 / CON\_MEM\_CHECK

	15	14	13	12	11	10	9	8	7	6	5	4	3	2	1	0
0	V	ersion	=0	0	1			Proc	ess ID	= 59			Pac	ket Ca	tegory	= 7
1	1	1						Sour	e Sequ	uence (	Count					
2							Pac	cket Le	ength =	= 19						
3							OB	ГTime	e (Seco	nds)						
4		OBT Time (seconds)														
5		OBT Time (fractional seconds)														
6		PUS     Chk     Spare     Packet Service type = 6														
7		F	Packet	Service	e subty	pe = 1	0					PAD	Field			
8			60	(dec) =	= 3C (h	ex)					Num	ber of	Blocks	s (=1)		
9							Start a	address	s MSW	/ (=0)						
10							Sta	art add	ress LS	SW						
11						Blo	ock Le	ngth (i	n 16 b	its wor	ds)					
12					CRO	C16 Va	alue of	the De	esignat	ed Me	mory A	Area				

### $ETM01702 \ / \ CON\_TEST\_RESP$

	15	14	13	12	11	10         9         8         7         6         5         4         3         2         1         0										0
0	V	ersion=	=0	0	1			Proc	ess ID	= 59			Pac	ket Ca	tegory	= 7
1	1	1						Sourc	e Sequ	uence (	Count					
2	Packet Length = 9															
3	OBT Time (Seconds)															
4							OB	T Time	e (seco	nds)						
5						OI	BT Tin	ne (fra	ctional	second	ls)					
6		PUS		Chk		Sp	are				Packe	t Servi	ce type	e = 17		
7		]	Packet	Servic	e subty	ype = 2	2					PAD	Field			

CONSERT Orbiter Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 98 / 102 Author : J.P. Goutail

Experiment User Manual

## ETM00606 / CON\_MEM\_DUMP

	15	14	13	12	11	10	9	8	7	6	5	4	3	2	1	0
0	V	ersion	=0	0	1			Proc	ess ID	= 59			Pac	ket Ca	tegory	= 9
1	1	1						Sour	ce Sequ	uence (	Count					
2							Р	acket I	Length	=						
3							OB	Г Time	e (Seco	nds)						
4							OB	T Time	e (seco	nds)						
5		OBT Time (fractional seconds)														
6		PUS		Chk		Sp	are				Pack	et Serv	rice typ	be = 6		
7		Packet Service subtype = 6 PAD Field														
8		60 (dec) = 3C (hex) Number of Blocks (=1)														
9							Start a	address	s MSW	/ (=0)						
10							Sta	art add	ress LS	SW						
11						Blo	ock Le	ngth (i	n 16 bi	its wor	ds)					
12																
							D	umped	Memo	ory						
?																

Orbiter

Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 99 / 102 Author : J.P. Goutail

### Experiment User Manual

## ETM02003 / CON\_SCI\_REP

	15	14	13	12	11	10	9	8	7	6	5	4	3	2	1	0
0	V	<sup>v</sup> ersior	1=0	0	1			Proc	ess ID	= 59			Pack	ket Cat	egory	= 12
1	1	1						Sourc	e Sequ	ience (	Count					
2							Pac	cket Le	ngth =	1041						
3							OF	BT Tim	e (Sec	onds)						
4							OI	BT Tin	ne (seco	onds)						
5						С	BT Ti	ime (fra	actiona	l secor	nds)					
6		PUS		Chk		Sp	are				Packe	t Servi	ice typ	e = 20		
7			Packe	et Servi	ce sub	type =	3					PAD	Field			
8		Sounding Start TIC (MSW)														
9	Sounding Start TIC (LSW)															
10	OCXO Temperature DIGI Board Temperature															
11	Present Sounding Number															
12			Preser	nt Gain	Contr	ol Wo	rd				00	CXO F	requer	ncy		
13							S	ignal I	Chann	el 0						
14							S	ignal I	Chann	el 1						
-																
267							Sig	gnal I (	Channe	1254						
268							Si	ignal Q	Chanı	nel 0						
269							Si	ignal Q	Chanı	nel 1						
-																
522							Sig	nal Q	Channe	el 254						
523									0							

0BBC	C007	0411	0000	00D4	A000	00 <u>14</u>	<u>03</u> 00	0000	D69A	AAAC	
		Len		Time		ΤŢ	уре	snds	strt	Temp	

CONSERT
Orbiter

Reference: RO-OCN-TN-3044Issue: 3.2Date: 09/11/04Page: 100 / 102Author : J.P. Goutail

### Experiment User Manual

#### Telecommand Packet Structure

	Packet Header														
							Pack	et ID							
V	<i>'ersion</i>	#	Type	Data				A	pplica	tion Pro	ocess I	D			
				Field			Pı	ocess l	D			P	Packet (	Categor	y
				Head											
				er Elag											
0	0	0	1	гаg 1	0	1	1	1	0	1	1				
0	U	0	1	1	U	Packe	t Seau	ence C	ontrol	T	I				
Segm	entati														
on F	lags	Sc	ource Pa	art				<b>I</b> _	Sequ	ience C	Count				
1	1														
	Packet Length = (236 + 4 + 2 ) - 1) Max													1	
Packet Data Field															
			-			Da	ta Fiel	d Head	ler						
Packe	et Utilis	sation	Chec		Ackno	wledge				Pac	ket Sei	rvice T	ype		
S	tandar	d	ksum Flag												
			1	0	0	0	1								
		I	Packet	Subtyp	e					1	Pad	Field	1	1	1
					S	ource l	Data (2	36 Byt	tes Ma	<u>x)</u>		1	1	1	1
			l		1	Pacl	ket Err	or Cor	ıtrol	1	1	1	1	1	1

Orbiter

Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 101 / 102 Author : J.P. Goutail

Experiment User Manual

## ETC19201 / MISSION\_TABLE

	15	14	13	12	11	10	9	8	7	6	5	4	3	2	1	0
0	V	ersion	=0	1	1			Proc	ess ID	= 59			Pacl	ket Cat	egory	= 12
1	1	1						Sour	ce Sequ	uence (	Count					
2							Pac	cket Le	ength =	32						
3		PUS		Chk		Sp	are				Packet	t Servi	ce type	e = 192		
4		]	Packet	Servic	e subt	ype = 1	1					PAD	Field			
5	$\frac{1}{1} Mission Table index (=1) + Spare (=0)}$															
6	Start TIC for Tuning phase MSW (= 65536 Tics = 107 sec)															
7	Start TIC for Tuning phase LSW (= 7706 Tics = 13 sec)															
8	Start TIC for Soundings MSW (= 0 Tics = 0 sec)															
9				S	tart TI	C for S	Soundi	ngs LS	SW (=	48218	Tics =	= 79 sec	c)			
10						Time s	step in	TIC (=	= 3052	Tics =	= 5 sec)	)				
11						Total	l numb	er of s	oundin	gs (= 1	1000)					
12			(	Clock I	nitial l	Freque	ncy Se	etting (	= 127)	+Moc	de Byte	e Settir	ng (= 0	)		
13					Min	imum	(=0) +	- Maxi	mum (*	= 31) A	Attenu	ation				
14					NBL /	AGC L	Level (=	=180)	+ NBL	Zero	Level	(=120)				
15					Pac	ket Er	ror C	ontrol	(Not to	ested b	y Con	sert)				

### ETC19202 / DIRECT\_TC

	15	14	13	12	11	10	9	8	7	6	5	4	3	2	1	0
0	V	ersion	=0	1	1			Proc	ess ID	= 59			Pack	tet Cat	egory	= 12
1	1	1						Sourc	e Sequ	ience (	Count					
2	Packet Length = 14															
3		PUS		Chk		Sp	are				Packet	Servi	e type	= 192		
4		]	Packet	Servic	e subty	ype = 2	2					PAD	Field			
5			DI	R_COI	MMA	ND					I	DIR_P	ARAM	1		
6					Pac	ket Er	ror C	ontrol	(Not te	ested b	y Con	sert)				

### **CONSERT Orbiter** Experiment User Manual

Reference: RO-OCN-TN-3044 Issue : 3.2 Date : 09/11/04 Page : 102 / 102 Author : J.P. Goutail

Other TC Packets ETC00305 / ENABLE\_HK ETC00306 / DISABLE\_HK ETC00602 / CON\_MEMO\_PATCH ETC00605 / CON\_MEMO\_DUMP\_RQ ETC00609 / CON\_MEMO\_CHECK\_RQ ETC00901 / ACCEPT\_TIME ETC01701 / PING\_TEST ETC02001 / ENABLE\_SC ETC02002 / DISABLE\_SC ETC25501 / RESET\_TM\_BUFFER