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GSFC STI PUBLIC DISCLOSURE EXPORT CONTROL CHECKLIST

Title: Deep Impact - Integration and Test Instrument - Document No: 564420

Instructions - This checklist is used for the disclosures of STI, and ITAR or EAR controlled information.

For STI (scientific and technical information) disclosures subject to NPG 2200.2 (Guidelines for Documentation, Approval, and Dissemination of STI), this checklist should be forwarded with the NASA Form 1676 (NASA Scientific and Technical Document Availability Authorization (DAA)). For all other ITAR (International Traffic in Arms Regulations, 22 CFR 120-130) and EAR (Export Administration Regulations, 15 CFR 730-774) disclosures, this checklist should be forwarded with a copy of the information to the GSFC Export Control Office for review/approval. Completion of this checklist should minimize delays in approving most requests.

Background Information

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Generally, the export of information pertaining to the design, development, production, manufacture, assembly, operation, repair, testing, maintenance or modification of defense articles (e.g., space flight hardware, ground tracking systems, launch vehicles to include sounding rockets and meteorological rockets, radiation hardened hardware and associated hardware and engineering units for these items) are controlled by the State Department under the ITAR. The export of information with respect to ground based sensors, detectors, high-speed computers, and "dual use" (military/commercial) technology items are controlled by the Commerce Department under the EAR. A complete listing of items covered by the ITAR and the EAR can be accessed at http://export.gsfc.nasa.gov. If the information intended for release falls within the above categories but fits into one or more of the following exemptions, the information may be released.

EXEMPTION I – Public domain information If your information is already in the public domain in its entirety through a non-NASA medium and/or through a NASA release previously approved by any NASA Export Control Office, the information is exempt. If the information falls into this category, you may attest that you are using this exemption by signing below. Print Name and Code Signature Date EXEMPTION II – Scientific data If your information pertains exclusively to the release of scientific data (e.g., data pertaining to studies of clouds, soil, vegetation, oceans, planets, stars, galaxies and the universe) without the disclosure of information pertaining to articles controlled by the ITAR or EAR, such as flight instruments, high-speed computers, or launch vehicles, the information is exempt. If the information falls into this category, you may attest that you are using this exemption by signing below.

Signature

Date

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GSFC STI PUBLIC DISCLOSURE EXPORT CONTROL CHECKLIST CONTINUED

EXEMPTION III – General purpose information								
If your information falls into the ITAR or EAR areas of concern as referenced above, but is offered at a general purpose or high level (e.g., poster briefs and overviews) where no detailed technical information (e.g., design, development, production, or manufacturing) pertaining to ITAR or EAR controlled items is offered, the information is exempt. If the information falls into this category you may attest that you are using this exemption by signing below.								
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exemption 125.4(b)(13) of the ITAR. This exert domain and does not require that the information only to agencies of the Federal Government. The receiving assurance that such a release is a respective use of this authority: The information may manufacture of an identified ITAR controlled is conjunction with other information in the public or defeat controlled U.S. technologies. All sign below) expressly indicate concurrence with the that you have met these criteria, you may attest Office will offer favorable consideration towar	temptions stated above, the information might be imption allows the release of ITAR controlled inform be published in order to qualify. Use of this explored the GSFC Export Control Office will approve this consible action. The following guideline has been not offer specific insight into the design, develop tem (reference paragraph 2 above) in sufficient did domain) to allow potential adversaries to replicatures of approval on NASA Form 1676 (and in responsible use of this exemption when it is required to the proving your request under this special exemption.	ermation into the public temption is afforded a exemption only after a established regarding ment, production, or letail (by itself or in eate the item, or exploit the signature blocks uested. If you determine SFC Export Control aption.						
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If your STI, ITAR, or EAR disclosure does not satisfy any of the above exemptions, please contact the GSFC Export Control Office for further clarification on the releasability of your information under the ITAR or EAR.

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This procedure describes the steps to be performed to test the throughput performance of the Medium Resolution Instrument (MRI) for Deep Impact. These tests will be performed as part of the overall Thermal Vacuum test described in Test Procedure 564426. Focus and boresight measurements are made at room temperature and are governed by Test Procedure 564422.

2 APPLICABLE DOCUMENTS

The following documents form a part of this procedure to the extent specified herein. Unless a specific issue or revision is listed, these documents shall be of the latest issue or revisions in effect at the time of the test. In the event of a conflict between this procedure and the reference documents, the contents of this procedure shall govern.

2.1 Deep Impact Program Documents / Drawings

559700	Instrument Platform Assembly
559698	Medium Resolution Instrument (MRI)
564304	Instrument Electronics Assembly, MRI
564307-501	Pre-Amp Clock Bias Assembly (MRI)
564422	Instrument Platform Alignment Procedure
564426	Instrument Platform Thermal Vacuum Procedure
564435	CSTOL: DI MRI Single Mode Dark Imaging
564436	CSTOL: DI MRI Single Mode Light Imaging
576704	CSTOL: DI MRI Three Frame Set Imaging
576705	CSTOL: DI MRI Five Frame Set Imaging
576700	CSTOL: DI MRI Power On
576701	CSTOL: DI MRI Power Off

3 TEST CONDITIONS AND REQUIREMENTS

3.1 Precautions

Failure to follow the procedures contained in this document, as well as SPS 1930 and SPS 3096, may result in damage to flight hardware.

3.1.1 ESDS Equipment. (If Applicable)

The test specimen contains electrostatic-discharge sensitive (ESDS) devices that may be exposed at the electrical interfaces. Therefore, it shall be handled per MIL-STD-1686 Class 1, as implemented by SPS 120211. The test specimen, the test operator (using wrist straps), and related electrical test equipment shall be connected to a common ground before any electrical connecting or disconnecting operations, and during the use of any electrical test equipment probes. The following requirements are important:

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- A. All personnel within 1 meter of the Instrument or Electronics, and all related test equipment shall be connected to a common ground at all time.
- B. An ionizing blower shall be in place and operating whenever static generators such as nylon sheeting or paper are within one meter of exposed electrical interfaces.
- C. Under no circumstances shall connections be made to the unit under test while power is applied.
- D. Test cables and equipment shall have all pins shorted to equal potential (or ground) prior to being connected to ESDS hardware. This requirement shall be met by using shorting devices on applicable connectors prior to mating.

3.1.2 Proof Load Certification

All equipment to be used for flight hardware lifts shall have current proof load certification. If certification paperwork does not exist, or is out of date, proof load testing must be performed, using a technique appropriate for the device in question. Immediately prior to use, each such device shall be visually inspected for damage. Any hardware exhibiting cracks, or any cable having parted strands, or any similar defects, shall be disposed of and replaced or documented on an MDR, regardless of its state of certification.

3.1.3 General Precautions for Connectors:

- A. Connector savers shall be used on all flight hardware electrical connectors as specified in the Deep Impact Connector Guidelines.
- B. Flight hardware electrical connectors shall be capped with ESD approved covers when they are not in use, to minimize contamination, and to prevent damage from electrostatic discharge.
- C. Before mating any connector, examine the connector to assure that there is no interference or visible contaminants at the pin or socket interface. Notify QA if connectors exhibit any problems.
- D. Extreme care shall be exercised should connectors need to be probed at any time, and the following precautions taken:
 - The probing pin shall be a mating pin or socket that is insulated to prevent shorting.
 - The probing pin shall be gold plated, or of a non-marring smooth surface.

3.2 Safety Requirements

In accordance with the Reliability, Parts and System Safety Handbook SP0031A-014. the System Safety Engineer or his delegate has final authority over safety provisions contained in this procedure, and in controlling any hazardous conditions which may arise during any operations performed in accordance with this procedure.

3.2.1 Responsibility for Safety.

All personnel are responsible for maintaining a safe work environment. The Test Conductor or cognizant operator shall assure that appropriate safe practices are implemented during these operations, and that operations are performed in a proper order.

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3.2.2 Test Readiness Reviews

Prior to the commencement of any activities detailed in this document, and again at each shift change, the Test Conductor or cognizant operator shall conduct a pretest briefing. Also, immediately prior to each hazardous sequence of operations, the Test Conductor or cognizant operator shall conduct a pre-task briefing. Both the pre-test and pre-task briefings shall include a discussion of:

- A. Test sequence, objectives, and equipment.
- B. Nature and location of the specific hazards to be encountered.
- C. Hazard controls, including protective equipment, safety boundaries, personnel access, etc.

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- D. Limitations on concurrent activity.
- E. Emergency instructions and response, and, when the situation warrants, the availability of emergency shutdown procedures.

3.2.3 Caution and Warning Notations

In this procedure follow these definitions:

- Caution: Operational step, etc., which if not adhered to or observed could result in damage to the equipment;
- Warning: Operational step, etc., when not adhered to or observed, could result in loss of life, personal injury, or exposure.

3.2.4 Mechanically Assisted Lift Hazard

Severe damage to personnel and flight hardware may result if lifting fixtures are excessively loaded. Lifting hardware shall never be used in a configuration that may apply loads greater than the working load that is clearly marked on each piece of lifting hardware. Never use lifting hardware that is not marked with a working load and proof test date. A properly certified crane operator shall control the lift.

3.3 **Cleanliness And Environment**

Except as provided herein, these operations shall be conducted in a Class 10,000 or better clean area as defined in FED-STD-209B, and as implemented by BATC process standard BPS 21.04. Cleanliness of the test units shall be established and maintained per BATC specification SPS 2780.

Standard laboratory conditions of atmospheric temperature (18 to 28°C), pressure (520 to 810 torr), and relative humidity (25 to 70 percent) are acceptable for the operations defined herein.

Controlled Redline Procedures 3.4

BATC DEEP IMPACT test procedure change control shall be implemented in compliance with DEEP IMPACT QWI.

A. Any testing of flight hardware shall be required to be done in accordance with a test procedure that has been formally released prior to the start of the applicable test. This

	
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means that test procedures must be released through BATC Engineering Document Control prior to use.

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- B. Once the test procedure is released, it may be changed in compliance with WI Test.4.3.005 Controlled Redline Test Procedure. Such changes may also be made during the test.
- C. Such changes must be approved by the following personnel:

Minor Changes: Test Engineer and Quality Engineer.

Major Changes: Test Engineer, Quality Engineer and Responsible Design Engineer and/or Systems Engineer (As determined by QE)

- D. The "AS-RUN COPY" of the test procedure containing such changes is controlled by the Test Engineer.
- E. Subsequent to completion of the applicable test, but before final buyoff of that test in the applicable Certification Log, the redline changes to the "AS- RUN COPY" test procedure shall be incorporated in a formal revision of the test procedure. (Note: At the discretion of the program the final Spacecraft Integration procedures may not be updated.)

3.5 Software Configuration

Prior to beginning any DEEP IMPACT instrument/spacecraft level test, the revision of all flight software and test software shall be recorded. The listing will include the software version numbers, tables and database version numbers. If any software is revised during the testing, record the revision information required identifying the software configuration used during each test operation.

4 Quality Assurance Provisions / Test Management Responsibilities

4.1 Quality Assurance Provisions

QA provisions operative during activities defined in this procedure are derived from the DEEP IMPACT Product Assurance (PA) Plan and the DEEP IMPACT Quality Work Instructions (QWI) document. These provisions, summarized below, identify the interfaces between QA and test personnel.

- A. The PEQA shall be notified, in advance, of performance of any activities described in this procedure. A QA Inspector shall be assigned to monitor those activities as is deemed necessary or appropriate, in accordance with the QWI. When required, QA shall notify the customer and/or government representatives, who may, at their option, monitor or witness the activity. The PEQA shall attend the "Ready-to-Test" meeting with assigned QA personnel, to assure QA support during the test, and to address and resolve outstanding QA items.
- B. The PEQA shall assure that the proper "as-run copy" of this procedure document has been prepared, and that the required Cert Log is in proper order. When all pre-test conditions have been met, the PEQA shall complete the "Ready-to-Test" entry in the applicable Cert Log.

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- D. Test Support provided by the QA Inspector shall be as follows:
 - Verify that calibration is current for test and measuring equipment in accordance with QSP 11.1 "Selection, Maintenance, and Control of Inspection, Measuring, and Test Equipment".
 - Verify that applicable lifting and handling GSE has been properly proof tested and tagged.
 - Assure that the proper change control practices are applied as described in section 3.4, which implements the applicable provisions of QSP 4.4 Design Changes.
 - Assist in assuring that the precautionary and safety requirements stated in sections 3.1 and 3.2 are met.
 - Assist in the inspection of test setups prior to application of power to a test specimen, and prior to any mechanism-assisted lifts or moves.
 - Prepare quality assurance test documents as applicable. In the Action Item List of the Certification Log, document any discrepancies or non-conformances noted during or after testing, and immediately notify the PEQA.
- E. At completion of the test, the PEQA/Inspector shall perform the following:
 - Review all test data for completeness, and to verify that all measurements are within tolerances.
 - Assure that the resulting test data fulfills the test requirements of the end item.
 - · Verify completion of any related action items and disposition of any related MDR's in the test Cert Log.
- F. Upon the occurrence of a test anomaly (any event that deviates from the planned procedures, exceeds normal variations, or generates unexpected data), operation of the test article shall be stopped immediately. All other test conditions and parameters shall be maintained (except as those conditions that may pose an immediate hazard). The Inspector and the Test Conductor shall review the anomaly. Minor, non-destructive, investigative troubleshooting that will not change the failure mode of the article under test, may be conducted by the Test Conductor, with PEQA concurrence. All troubleshooting shall be performed in accordance with QSP 13.1 Control of Nonconforming Product and WI PA.13.1.004 Test Anomaly Work Sheet (TAWS). If the anomaly is found to result from human error or test equipment problems that have not affected the test article, corrective action shall be taken and testing may continue. The PEQA will check the "continue test" box, the approval box, and sign the TAWS along with the Responsible Engineer.
- G. If a test anomaly cannot be resolved as described above, the PEQA shall be notified, and an appropriate entry shall be made in the Action Item List (AIL) portion of the Certification

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Log. The PEQA shall prepare a Material Discrepancy Report (MDR) and convene the Material Review Board (MRB) for action as specified in WI PA.13.1.002 "Completion of Material_Discrepancy Report". The MRB convened to resolve an anomaly that occurred during testing shall include a representative of the project test group. Testing of the failed item shall then not continue without prior authorization from the MRB.

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4.2 Test Management Responsibilities

4.2.1 Integration And Test Manager

The Integration and Test Manager shall have responsibility for the following:

- a. Delegate responsibility to conduct the SIM Alignment and Test Procedure or portions of this procedure to qualified personnel.
- b. Release of the Alignment and Test Procedure.
- c. Release of all required procedures, drawings, E.O.'s and other documents.
- d. Approve exceptions to requirements of this procedure, including facility ambient requirements.
- e. Availability and certification of required GSE.
- f. Approval on the final completion of the SIM Alignment and Test Procedure

4.2.2 Test Conductor

The test conductor (T/C) as delegated by the Integration and Test Manager is assigned shift responsibility for the following:

- a. Verify that required flight hardware and GSE are available and certified for integration and alignment as described in this procedure.
- b. Verify that GSE and other test equipment is available and calibrated where applicable and that calibration will not expire during performance of this procedure.
- c. Verify that required documentation is released, correct and complete. Documentation will include:
 - A copy of the latest revision of the Integration and Alignment Procedure
 - The Certification Log for the Instrument
 - The Assembly drawing for the FPA integration to the Telescope Assembly
 - All necessary prints, E.O.'s and other documents

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- d. Verify that setup, integration and alignment are in accordance with the requirements of this procedure, including facility ambient conditions. Setup shall be verified by an independent observer and recorded in the Certification Loa.
- e. Collect, identify and store all raw data generated during the procedure. Data storage will be in the program SER files and referenced in the Certification Log.
- f. Responsible for sign off, on completion, all integration and alignment process steps, torque values and / or other measured and recorded values as specified in this procedure in the Certification Log. The person who performed the operation shall perform sign off.
- g. Conduct the integration and alignment in a safe manner.
- h. Notify the Integration and Test Manager and QA monitor or PEQA if out of tolerance conditions occur. Flag out of tolerance conditions in the Certification Log.
- i. Supervise the troubleshooting and rework as required.
- j. Maintain a record in the Certification Log of all action items and ensure that all items are closed prior to final sign-off of the Certification Log.
- k. Notify the Integration and Test Manager and QA monitor or PEQA of completion of this procedure for data review and sign-off in the certification log.

The test conductor has the authority to change, in writing, the sequence of events of this procedure, during the conduct of this procedure, to facilitate availability of certain test equipment, flight hardware or personnel. The T/C is responsible for certifying that such deviations do not jeopardize the intent or the data integrity of this procedure.

The test conductor has the authority to substitute equivalent test equipment for those items called out in Section 6 of this procedure. The T/C is responsible for certifying that the substituted items are up-to-date in terms of their certification and that these substitutions do not jeopardizing the intent nor the data integrity of this procedure.

TEST DESCRIPTION

This procedure describes three tests, which will be run at different thermal conditions, as specified in the Instrument Platform Thermal Vacuum Test Procedure, 564426.

- 1) A set of CCD dark current measurements taken as the CCD cools down to its operating
- 2) An optional test is a set of images of a bright spot followed immediately by images taken without the bright spot to determine the level of latent charge remaining in the CCD. This will be done in coordination with steps in the alignment and focus test procedure 564422

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3) Throughput calibration using an integrating sphere. This is used to establish a calibration (using both the previous sphere calibration, and a reference photodiode), establish system flatfield images, determine system linearity, and test system gain using the gain-variance technique.

TEST ARTICLE CONFIGURATION

The MRI shall be configured as described in the as-run copy of 564426, Instrument Platform Thermal Vacuum Procedure

7 REQUIRED TEST EQUIPMENT

- 24" LabSphere integrating sphere system including sphere, calibrated lamp and lamp
- 65" LabSphere integrating sphere system including sphere, calibrated lamp and lamp controllers
- Optical Test Station #2, CW light source, ND2 filter
- Calibrated photodiode and amplifier

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Inspect all electrical and mechanical setup

Confirm that last functional test shows that system is operating properly

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Confirm that QA has given OK to test in Cert Log

For each section below, if not explicitly stated, use the following CSTOL procedures to turn instrument electronics power on and off:

Power ON: TP576700 Power OFF: TP576701

When powering electronics on or off, notify the personnel monitoring the chamber thermal conditions so they can compensate for change in power.

Dark Current Measurements During CCD Cooldown 8.1

The cooldown of the MRI CCD offers a unique opportunity to measure the CCD dark current vs. temperature for a range of temperatures expected in flight. During the steps to cool the CCD specified in Test Procedure 564426, conduct the following steps as needed to gather images.

- 1. Cover the window of the TV chamber with either dark cloth or Al foil (or both) to greatly Both on main wind blich cloth on ZnSe reduce stray light
- 2. Power on the MRI instrument electronics using CSTOL script TP576700. Alert test operator that instrument electronics are powered on.



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- 4. Repeat the following steps as needed. \checkmark
- a) Record the current CCD temperature in the image log, as measured by the test temperature monitor

3. Start the MRI VIS Single Mode Dark CSTOL script, TP564435 (Single Mode Dark) -

- b) Use Figure 8-1 below to determine the approximate integration delay time needed to produce at least 5 DN of signal at the measured CCD temperature
- c) Collect two to four (2-4) MODE 1 images with 0 msec integration delay
- d) Collect two to four (2 4) MODE 1 images with the integration delay calculated in the previous step, using CSTOL script TP564435 (shutter closed)
- e) Using DIVe (or other analysis tool) determine the dark current by measuring average signals in the image area and the serial overclocked region (columns 0-7 or 1015-1023)
- f) At the discretion of the test conductor, take images of other modes (2-9) during cooldown using same integration delay time calculations
- 5. Power down instrument electronics using CSTOL script TP576701, and alert test operator that instrument electronics are powered off.

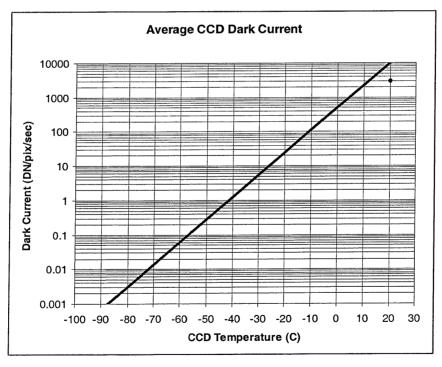


Figure 8-1 Average Dark Current For Deep Impact CCDs.

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8.2 Latent Image and Large Sphere Tests (Optional)

В	.2.	1	La	te	nt	lm	aa	е

- 1. During alignment testing of MRI using optical test station #2 and at the discretion of test director and systems engineering, perform the following \
- 2. Place the knife edge target in the target stage of optical test station #2
- 3. Remove the end of the fiber optic feed from the flash lamp and insert in the CW lamp
- 4. Turn on CW lamp and place diffuser behind knife edge target with the fiber optic feed shining through an ND-2 filter and onto the diffuser — place ND2 in front of
- 5. Collect MODE 1 image, 100 msec integration delay, using CSTOL script TP564436 and use DIVe to determine the signal level of the lit portion of the knife edge target
- 6. Adjust the output of the CW lamp and repeat the previous step until the average signal in the lit portion is not saturated
- 7. Remove the filter.
- 8. Using CSTOL script TP564436, set up to collect a series of 16 MODE 1 images with 100 msec integration delay and follow the steps below. Image count can be found in the Detector Ops telemetry window with mnemonic IIIDOVISEXCC
 - a. Place a light block in front of the chamber window
 - b. Start the image sequence
 - c. When the image count reaches 2 remove the light block
 - d. When the image count reaches 6 replace the light block and let the sequence finish
- 9. Examine the images using DIVe and determine that there are at least 3 images with a saturated CW lamp filament image. If not repeat the image sequence and light block steps -repeat in mode 6 with 64 images
- 10. With the same light setup as in step 8, collect 4 MODE 1 images with 10 sec integration delays using CSTOL script TP564435, single mode dark to test the shutter blocking.

8.2.2 Large Sphere

While the large (65") integrating sphere is being used for calibration of HRI, and at the discretion of the test director and systems engineering, perform any or all of the following steps

1. Move the instrument platform thermal vacuum translation stage so that the MRI is addressing the 16" fused silica chamber window and therefore the 65" integrating sphere. v

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2.	If the integrating sphere is not lit by a single internal lamp, follow the steps in 564419 to get to that state Lamp E
3.	Confirm that the MRI filter wheel is at position 1. If not move the filter wheel to position one by issuing the OASIS command "CMD IMF VIS_FW_ABS WITH FILTER_NUM 1"
4.	Collect 8 MODE 1,2,3,5 and 9 images, 0 msec integration delay using CSTOL procedure TP564435, single mode dark. These are bias frames.
5.	Collect 8 MODE 1,2,3,5 and 9 images, looking at lit integrating, sphere, with 0 msec integration delay using CSTOL procedure TP564436, single mode light
6.	Use DIVe to determine the average signal level in each quadrant. Calculate the integration delay time that will saturate the detector (14,000 DN), using 8.5 msec exposure time for 0 msec integration delay. Use that calculation to determine a set of integration delays that will produce up to 120% of detector saturation.
8. 9.	Collect 8 MODE 1 images with the exposures determined in the previous step Move the filter wheel to position 6 (clear) using the OASIS command "CMD IMF VIS_FW_ABS_WITH_FILTER_NUM 6" Repeat steps 4 through 7 at filter positions 6
	Repeat steps 4 through 7 at filter positions 4 and 5 Repeat steps 4 through 7 at filter positions, 7, 8 and 9, with integration delay times that produce a signal at least ¼ full well
13.	Install UDT photometer in front of 65" integrating sphere Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with UDT and Window"
15.	Cover the "Rambo" chamber window with black cloth or low reflectivity material Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with UDT, Window Covered"
	Remove the UDT calibrated photodiode
17.	Use a Photometer Worksheet to collect photometer measurements of the integrating

17. Use a Photometer Worksheet to collect sphere with "Lighting Conditions" labeled "Lamp without UDT, Window Covered" >

18. Remove the dark cloth from chamber window

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8.3 MRI Flatfield Source Imaging

Tests at room temperature indicate that the un-attenuated quartz-halogen source feeding the 24" LabSphere will saturate the CCD for very short exposure times. Thus the calibration and linearity tests require that the variable aperture (attenuator), which is part of the LabSphere, will need to be used to bring the source to a level that will allow for on-scale imaging. The following steps allow a determination of the final source intensity used to calibrate the MRI, compared to the un-attenuated source intensity. Photometer readings will be collected on sheets labeled Instrument Platform Calibration Photometer Worksheet included in the Appendix. Make as many copies as needed. This will be shortened to "Photometer Worksheet" in the procedure steps.

8.3.1 Cold Dark Measurements

When the MRI structure and the MRI electronics have reached their expected operating temperature, as defined in test procedure 564426, conduct the following steps.

1. Record CCD temperature as read from the test temperature monitor $\frac{-97.34}{2}$

2. If not already on, power on the instrument electronics using CSTOL procedure TP576702

3. Cover the thermal vacuum chamber window with dark cloth or Al foil (or both)

4. Start CSTOL procedure TP564435 (Single Mode Dark)

5. Collect sixteen to thirty-two (16 – 32) image in each of 9 MODEs with the minimum

24 (00)

integration delay time (0 msec for modes 1, 2, 3, 5 and 9; 4 msec for modes 4, 6, 7 and 8)

mod 9 +8

6. Determine minimum exposure time to produce at least 5 DN of signal from dark current, either from Figure 8-1 or from extrapolated data from section 8.1 W/A 5DN > 1hour

7. Collect sixteen to thirty-two (16 - 32) images in each of 9 MODEs with integration delay equal to the exposure time calculated in the previous step

8. Power down instrument electronics using CSTOL script TP576703, and alert test operator that instrument electronics are powered off.

9. Uncover thermal vacuum chamber window \checkmark

10. Record CCD temperature as read from the test temperature monitor $\frac{-9.8}{\text{Also see troops}}$.

8.3.2 Installing the Integrating Sphere

 Place the front aperture of the 24" LabSphere integrating sphere in front of the chamber window so that the center matches the MRI boresight outside the chamber within 1 inch.

The approximate locations of each component is shown in Figure 8-2. Note used 20" fold flat on option test station 2

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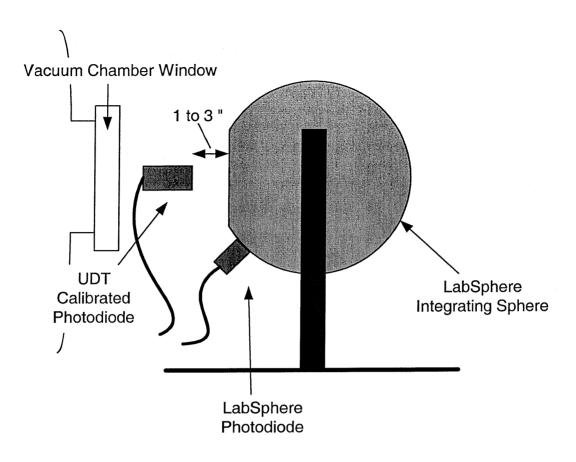


Figure 8-2 Integrating Sphere Mechanical Setup

- 2. Connect COM port of a PC computer with LabSphere USS-2000V software to LabSphere control module (Model SC5500, port 1 using serial cable to null modem cable.
- 3. Start USS-2000V software on laptop -
- 4. Turn on Labsphere control module, interface module and lamp power supply C >
- 5. Confirm that the Labsphere photodiode is operating
- 6. Confirm that variable aperture (VA) control is working
- 7. Set the VA to 255 (full open) for section 8.3.3

8.3.3 Ambient signals

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- 1. Record CCD temperature as read from the test temperature monitor ___96.5, __96.4
- 2. If not already on, power on the MRI electronics using CSTOL script TP564700 N/A on

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			DWG NO.	564420	sн 16	REV	
1	Collect eight to sixteen (8-16) images in a	all MC	ODFe with	the minimum ir	oteoration	delay ((า
⁴.	msec for MODEs 1, 2, 3, 5 and 9, 4 msec for						,
	bias frame measurements always done	OI UL	1612) using	J OS I OF Souhr	. I ГОО ТТ О	5 101	
_				U-1 intogratio	boro	'n io	
5.	not on (is on, UA=0 - this also					lamp is	
•	·	_	•		•	1.111	
3.	0 0						-
	overhead lights above the sphere be turned						
	so that the skylights do not introduce variab						S
	needed to reduce ambient light as needed	-		-			Foll
	Collect eight images in MODE 1 with 1 sec						5 K
	as dark images (USE images from Sector Collect eight to sixteen $(8-16)$ images in a	con (3.3.1)	~ 10 sec A	3/2/03	CB	.7
3.	Collect eight to sixteen (8 – 16) images in a	all MC	DDEs with	ြွှားနှင့် integratio	on delay u	sing	J
	CSTOL script TP564436 as an ambient ligh		•				
€.	Move to each filter and collect eight (8) imag	ges i ،	in MODE 1	with 10 sec in	tegration (delay 🗸	
	using CSTOL script TP564436. If any of the		•		1.0	t with	
	shorter exposure until the images are within					10 nn	T
	Install the calibrated photodiode (UDT) in fro						f 103
	Use a Photometer Worksheet to collect amb						۱۵.
	Conditions" labeled "Ambient"			162.	,3K		
2.	Record CCD temperature as read from the	test '	temperatur	re monitor <u>-95</u>	.1c, -95.	,OC_	
	.4 Source Measurements						
	Confirm that the variable aperture is set to 2		-				
	Turn on the calibrated quartz halogen lamp			,			
	sphere to the calibrated current level of 6.25	50 ar	nps and all	low at least 30	minutes to	o bilt	
;	stabilize Record lamp run time from LabSphere contr Record the stabilized current 6.250 and			and Hark	Slample	2.3/6	103
.	Record lamp run time from LabSphere contr	roller	r at power o	on Seculion	K Inni	j.,.	-
.	Record the stabilized current 6.250 and	l volt	.age _ <i>N / J</i>	AV			
. (Use Photometer Worksheet to collect photor	mete	er measure	ments of the in	ntegrating	sphere	i
١	with "Lighting Conditions" labeled "Lamp witl	ih UE	OT and Wir	ndow"			:
. (Cover the "Rambo" chamber window with bl	lack	cloth or low	ง reflectivity ma	aterial 🗸		
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		<u> </u>	DWG NO.	564	420	SH	17 F	EV -
7.	Use a Photometer Worksheet to collect ph sphere with "Lighting Conditions" labeled '						grating	I/N a vot
8.	Remove the UDT calibrated photodiode							(b
9.	Use a Photometer Worksheet to collect ph	otome	ter measu	ureme	ents of the	integ	grating	
	sphere with "Lighting Conditions" labeled "	'Lamp	without U	DT, V	Vindow Co	overe	d" —	
10.	Remove the dark cloth from chamber wind	low						
11.	Set the LabSphere variable attenuator to o	closed	by enterin	ng 0 ir	the USS	-2000	V "VA" f	ield 🌽
12.	Set the variable attenuator to 20 10						_	1.10
13.	Re-install the UDT photometer in front of the	he inte	grating sp	here			N/A	NO
14.	Use a Photometer Worksheet to collect ph	otome	ter measu	ıreme	nts of the	integ	rating	
	sphere with "Lighting Conditions" labeled "	Lamp	with UDT	and V	Vindow"			
15.	Remove the UDT photometer							
16.	Use a Photometer Worksheet to collect ph	otome	ter measu	ıreme	nts of the	integ	rating	
	sphere with "Lighting Conditions" labeled "	Lamp	with Wind	ow, n	o UDT" L			
	.5 Clear Filter Measurements	-						
1.	Verify that the MRI filter wheel is set at pos	sition 1	(Clear Fil	lter). I	f not mov	e to p	osition 1	by
	issuing OASIS command "CMD IMF VIS_	_FW_AI	BS WITH	FILT	ER_NUM	1" ar	nd confir	m
	position by reading Mechanism telemetry p	oage _						
2.	Collect MODE 1 images using CSTOL scri	pt TP5	64436 an	id ana	lyze in re	al tim	e using	
	DIVe to find an integration delay that produ	ıces in	nages with	n sign	al level th	at is t	oetween	
	4000 and 5000 DN above the serial overclo	ocked	region sig	ınal (fı	ull well is	~140	00 DN)	
3.	Use Table 8-1 below to spread 20 to 25 int	egratio	on delay ti	mes f	rom 0 ms	ec to	that nee	ded
1	to produce 120% of the nominal CCD full w	vell (45	50,000 e-,	15,00	00 DN) $ u$			
4.	For each integration delay entered in Table	8-1 c	ollect eigh	nt to tv	venty-foui	r (8 –	24) MOI	DE
	1 images using CSTOL script TP564436 (s	ingle r	node light	t) Se	ee logshe	uts		
	Collect sixteen (16) MODE 2, 3, 5 and 9 in				_		6 with a	า
i	ntegration delay at the midpoint of the dela	ays list	ed in Tabl	e 8-1	10	sec		
	Collect sixteen (16) MODE 4, 6, 7, and 8 in							
r	nsec integration delay (minimum for these	mode	s). _A	Iso	10sec	pl.		
			V.	,				
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- 7. Move the MRI filter wheel to position 6 by issuing OASIS command "CMD IMF VIS_FW_ABS WITH FILTER_NUM 6" and confirm positions by reading from mechanism telemetry page
- 8. For each integration delay entered in Table 8-1 collect eight to twenty-four (8 24) MODE 1 images using CSTOL script TP564436 (single mode light) \vee
- 9. Collect sixteen (16) MODE 2, 3, 5 and 9 images using CSTOL script TP564436 with an integration delay at the midpoint of the delays listed in Table 8-1 1000ms
- 10. Collect sixteen (16) MODE 4, 6, 7, and 8 images using CSTOL script TP564436 with A msec integration delay (minimum for these modes)
- 11. Collect eight (8) MODE 1 images using CSTOL script TP564435 with 0 msec integration delay \
- 12. Collect eight (8) MODE 1 images using CSTOL script TP564435 with the longest integration delay listed in Table 8-1 for a shutter blocking test

Table 8-1 Clear Filter Exposure Times

	him -			Apoodio iiii		4	, , 、
#	T_exp (msec)	% Full Well	#	T_exp (ms	ec)	% Full	Well
1	0 ~		15	0 [16]	<u> </u>	1900	[8] ~
2	15		16	1 [16]		2000	[8]
3	オレン		17	2 [16]		2100	[8]
1	229 /		18	3 [16]	/	2200	[8]
5	G90		19	5 [16]		2300	[8]
6	2267		20	10 [6]		2400	[8]
7	1500		21	30 Li6	コレ		
8	15000		22	100 [8			
9	30000		23	300 [8	74		
10	35000		24		<u>\$</u>] 4	_	
11	22500		25				
12			26	1250 B)V		
13	10000		27		37~	/	
14			28	1750 E	874		

8.3.6 Bandpass Filter Measurements

- 1. For the 100 nm bandpass filters (4 & 5) perform the following steps
 - a. Move the filter wheel to the next filter positions by issuing the OASIS command "CMD IMF VIS_FW_ABS WITH FILTER NUM N" where N is the desired filter position

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- b. Collect a MODE 1 image with 10 msec integration delay using CSTOL script TP564436 and analyze in real time using DIVe to find the average signal level.
- c. Use Table 8-2 below to spread up to 12 integration delay times from 0 msec to that needed to produce 120% of the nominal CCD full well (450,000 e-, 15,000 DN)
- d. Record CCD temperature as read from the test temperature monitor 186.5
- e. For each integration delay entered in the table collect eight to twenty-four (8-24) MODE 1 images using CSTOL script TP564436 (single mode light)
- f. Collect sixteen (16) images of MODEs 2 through 9 CSTOL script TP564436 with an integration delay at the midpoint of the delays listed in Table 8-2
- g. Repeat steps a. through f. for the next filter
- h. Re-install the UDT calibrated photodiode at the aperture of the integrating sphere at the same place as previously installed

 N/Around
- i. Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with UDT and Window"
- j. Remove the UDT photometer
- k. Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with Window, no UDT" 7275.
- I. Record CCD temperature as read from the test temperature monitor 186.5K

Table 8-2 100 nm Bandpass Filter Integration Delay Times

	#	Filter 4	Filter	⁻ 5
		T_delay (msec)	T_delay	(msec)
	1	0/16	O	10
	2	3/16	G	IG
	3	10 6	20	8
	4	30 / 8	60	6
	5	100/8	200	6
	6	300 /8	600	6
	7	1000 6	2000	8
	8	1500 A	3000	4
Λ1.	9	2000 /4	4000	4
Sull -	10	2300/4	4400	4
June 3	11	2200 A	6600	4
1-2×500/	12			
1-0	13			
	14			

VA = 25

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- 2. Set the variable aperture to 0 \
- 3. Set the variable aperture to 115
- 4. For the visible band gas photometry filters (2 & 3) perform the following steps
 - a. Move the filter wheel to position the next filter position by issuing the OASIS command "CMD IMF VIS_FW_ABS WITH FILTER_NUM 2"
 - b. Record a MODE 1 image using CSTOL script TP564436, with \sec integration delay.
 - c. Use DIVe to determine the average signal level and determine the integration delay time needed to produce 120% of the nominal CCD full well (450,000 e-, 15,000 DN). If the 120% integration delay is significantly longer than 1 minute adjust the variable aperture and repeat the MODE 1 image until the integration delay is ~ 1 minute or less, or the variable aperture is open. Record variable aperture setting used
 - d. Use Table 8-3 to spread up to 12 integration delay times from 0 msec to that needed to produce 120% of the nominal CCD full well (450,000 e-, 15,000 DN). ν
 - e. Record CCD temperature as read from the test temperature monitor 186.35 K
 - f. Re-install the UDT calibrated photodiode at the aperture of the integrating sphere at the same place as previously installed
 - g. Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with UDT and Window"
 - h. Remove the UDT photometer
 - i. Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with Window, no UDT"
 - j. For each integration delay entered in the table collect eight to twenty-four (8 24) MODE 1 images using CSTOL script TP564436 (single mode light)
 - k. Collect sixteen (16) images of MODEs 2 through 9 CSTOL script TP564436 with an integration delay at the midpoint of the delays listed in Table 8-3
 - I. Move filter wheel to the next filter position by issuing an OASIS command "CMD IMF VIS_FW_ABS WITH FILTER_NUM N" where N is the desired filter number
 - m. Collect one MODE 1 images with integration delay of 2 sec >

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- n. Use DIVe to determine the average signal level and determine the integration delay time needed to produce 120% of the nominal CCD full well (450,000 e-, 15,000 DN).
- o. Repeat steps j. and k.
- p. Re-install the UDT calibrated photodiode at the aperture of the integrating sphere at the same place as previously installed N/A
- q. Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with UDT and Window"
- r. Remove the UDT photometer
- s. Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with Window, no UDT"
- t. Record CCD temperature as read from the test temperature monitor 18.5 K

Table 8-3 Visible Gas Photometry Bandpass Filter Integration Delay Times

		ш	Filter 2	Filter	2	7
		#		1		
		<u> </u>	T_delay (msec)	T_delay (r		4
		1	0	0	LG	
		2	5 v	10	16]
		3	10	RO	8	
		4	20 /	60	6	
		5	30 -	100	6	
		6	50	300	8	
		7	100	600	8	
		8	150	2000	g	
		9	300	6000	4	
		10	1000	20000	(
		11	3000	3000 U		- A.
		12	10000	42000		- Pull u
,		13	15000	50000	J	4-1.2X
	->	14	21000			
	1.2	X->	25000			

5. Set the variable aperture to 255 (full open)

6. For the UV gas photometry bandpass filters (7, 8 & 9) Perform the following steps

a. Move the filter wheel to position 9 by issuing the OASIS command "CMD IMF VIS_FW_ABS WITH FILTER_NUM 9"

1 30 sec

10 see

b. Cover the thermal vacuum chamber window and collect a MODE 1 image with 30 sec integration delay using CSTOL script TP564435 as a dark current measurement \checkmark

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- c. Uncover the thermal vacuum window and collect a MODE 1 image with 30 sec integration delay using CSTOL script TP564436
- d. Analyze the difference of the two images using DIVe, and calculate exposure time needed to produce full well, 14000 DN. If this time is significantly longer than 600 sec (10 minutes) then a second lamp may be required perform steps in Section 8.3.7, otherwise continue with these steps Yes Use 3 lamps
- e. Use Table 8-4 below to spread 10 exposure times from 0 msec to full well
- f. Install the full spectrum UDT photodiode in front off the integrating sphere
- g. Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with Window and UDT"
- h. Install UDT UV photodiode (model 222) in front of integrating sphere
- i. Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with Window and UDT"
- j. Remove the UDT photometer
- k. Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with Window, no UDT"
- I. For each integration delay time listed in Table 8-4 collect 2 to 8 MODE 1 images using CSTOL script TP5744361
- m. At an intermediate exposure time in Table 8-4 collect 8 to 12 MODE 2 through 9 images
- n. Cover the thermal vacuum chamber window with black cloth or Al foil (or both) and take a single MODE 1 image using CSTOL script TP576435 for the longest exposure time used in Table 8-4 for a dark current measurement

o. Repeat the previous two steps for filters 7 and 8. Move to filter position using OASIS command "CMD IMF VIS_FW_ABS WITH FILTER_NUM N", where N is the desired filter number. Begin with short exposures and work out in real time the exposure to achieve full well.

- p. Re-install the full spectrum UDT photodiode in front of the integrating sphere
- q. Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with Window and UDT"

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- Install UDT UV photodiode (model 222) in front of integrating sphere nou DT
- s. Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with Window and UDT"
- Remove the UDT photometer
- u. Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with Window, no UDT

Table 8-4 UV Gas Photometry Bandpass Filter Integration Delay Times 3 mse V16

3 lamps VA=255

#	Filter 7	Filter 8	Filter 9
	T_delay (msec)	T_delay (msec)	T_delay (msec)
1	0 916	0 ~16	0 0
2	1 \$16	1 16	4 10
3	2 4	2/16	2 30
4	5 (5~ 16	\$ 100
5	10 8	10 /16	#1 300
6	30 g	30 /10	1000
7	100 8	100 / 10	\$10 3000
8	250 g	300 V B	119 \$ 10000
9	500 0	1000 - 8	1\$ p \$ 3000t
10	1000 +	30000	110000 100000
11	2500 F	10000	30000
12	5000 4	33,000 40,000	4

~ 1900 DN/500 / 8600 +

8.3.7 Additional lamps for UV measurements

If lamp C in the 24" integrating sphere does not produce enough UV signal for the 309 nm filter, additional lamps may be used. Complete the following steps to take UV filter images.

- 1. Complete the steps in 8.4, Source Verification
- 2. Turn on lamp power supply controller for an internal integrating sphere lamp 1
- 3. Verify that the integrating sphere fan is operating L
- 4. Check that the lamp current is properly set \(\mathcal{l} \)
- 5. Turn on internal lamp and allow 15 minutes to stabilize ν
- 6. Cover thermal vacuum chamber window with black cloth or Al foil (or both)
- 7. Collect a MODE 1 image with 30 sec integration delay using CSTOL script TP564435 as a dark current measurement
- 8. Uncover thermal vacuum chamber window to view lit integrating sphere

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- 9. Collect a MODE 1 image with 30 sec integration delay using CSTOL script TP564436
- 10. Analyze the difference of the two images using DIVe, and calculate exposure time needed to produce ¼ full well, 3000 to 4000 DN. If this time is significantly longer than 600 sec (10 minutes) then another lamp will be required. Repeat the previous steps for a second internal lamp
- 11. Return to step 3.-e. of section 8.3.6

8.4 Source Verification

With the single calibrated lamp powered on at the integrating sphere, perform the following steps

- 1. Return variable attenuator to full open, 255, using USS-2060V software
- 2. Re-install the UDT calibrated photodiode at the aperture of the integrating sphere at the same place as previously installed
- 3. Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with UDT and Window, VA = 255"
- 4. Remove the UDT photometer
- 5. Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with Window, no UDT, VA = 255"
- 6. Move the variable attenuator to 0, full closed
- 7. Move the variable attenuator to 1/5
- 8. Re-install the UDT calibrated/photodiode at the aperture of the integrating sphere at the same place as previously installed
- 9. Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with UDT and Window, VA=75"
- 10. Remove the UDT photometer
- 11. Use a Photometer Worksheet to collect photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Lamp with Window, no UDT, VA = 75"
- 12. Record CCD temperature as read from the test temperature monitor _
- 13. Either return to section 8.3.7, or continue to section 8.5, Lamp Power Off

U/A See calibration data from March 21, 2003



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8.5 Lamp Power Off

1. Turn off power to lamp C using either UVS-2000 software control or from the front panel of the Labsphere Lamp power supply

2. Record lamp run time from LabSphere controller at power off

3. Use a Photometer Worksheet to collect Labsphere photometer measurements of the integrating sphere with "Lighting Conditions" labeled "Ambient, no UDT" >

8.6 Visible Stimulator Test

To test the operation of the LED that will be used as a visible stimulator conduct the following steps

- 1. Confirm that the primary/secondary mirror decontamination heater circuit, side A, is connected to 35 V power supply \
- 2. Disconnect leads to the heater circuit and set up power supply to provide 28 V, and turn off power supply V
- 3. Reconnect the beamsplitter decon. heater circuit leads to the power supply
- 4. If not already on, power on the MRI instrument electronics using CSTOL script TP576700
- 5. Collect four MODE 1 images with 100 msec integration delay, using CSTOL script TP564435 (Single Mode Dark)
- 6. Turn on power supply \checkmark
- 7. Collect four MODE 1 images with 100 msec integration delay, using CSTOL script TP564435 ∨
- 8. Turn off power supply \checkmark
- 9. Repeat steps 1 through 8 with the power supply set to 26V and 35V
- 10. Turn off MRI instrument electronics using CSTOL script TP576701

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DATA PRODUCTS & WORKSHEETS

The following table lists the calibration products and the associated procedure and section where the data are collected to produce the needed results

Product	Document	Section	Comments
Plate Scale	564422	8.1	Survey point source with theodolite
IFOV	564422	8.1	
FOV	564422 &	8.1	1008 active pixels x IFOV
	564446		
PSF	564422	8.2	Knife Edge and/or Pinhole Tests
MTF	564422	8.2	«
Noise	564420	8.3.5 &	Gain Variance
		8.3.6	
Readout Smear	564420	8.2.1	Latent Image Tests
Gain	564420	8.3.5	Gain Variance. Each Quad independently
Linearity	564420	8.3.5	Gain Variance, per filter
Flat Field	564420	8.3.5 &	Per filter
		8.3.6	
Bad Pixel Maps	564420	8.3.5	Hot and Dead pixels
Radiometric	564420	8.3.5 &	Calibrated lamp and sphere
		8.3.6	
Stellar Radiometry			Combine Radiometric and PSF results
MSSR 5.2.13.2-4			,

The following sets of images will be produced for each subsection of section 8

The following sets of images will be produced for each subsection of section of				
Subsection	Mode	# Images	Comments	
8.1	1 (1-9)	40 to 160	Based on HRI SIM test Aug. 25, 2002	
8.2.1	1	32	Latent Image	
8.2.2	1-9	360	Large Sphere	
8.3.1	1-9	576	Dark	
8.3.3	1-9	361	Ambient	
8.3.5	1-9	1477	Clear Filter	
8.3.6.1	1-9	833	100 nm bandpass filters	
8.3.6.4	1-9	833	Visible Gas Emission Filters	
8.3.6.6	1-9	530	UV Gas Emission Filters	
	Total	5130		

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Instrument	Platform	Calibration	Photometer	Worksheet
monument	Flationin	Calibration	FIIOLOINELEI	AAOLVQUICCE

Procedure & Section	564420		Instrument	MRI
Date & Time	0/27/03	3 15:10	Var. Atten.	0
Lighting Conditions	Ar	nbient	every - 1	ulfson
UDT Photometer			2	,
Readings				
Average Reading				
Labsphere Photometer				
Readings x 4	-8			
1.145	1.180	1.195	6194	(,173
1.19	1.169	1.186	1,193	1.181
Average Reading	-			
Comments				

Procedure & Section Date & Time Lighting Conditions UDT Photometer	2/27/0		Instrument Var. Atten.	mrt 20
Readings				
Average Reading		_		
Labsphere Photometer Readings	· · · 8			
1.876	1,969	1.875	1.463	1.869
687	1.454	1.863	11871	1,866
Average Reading		,		
Comments				
	pre -	30° Slats	_newr use .0:00 -san	VA-20
	Sec	post a z	0:00 -san	uday

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Instrument Platform Calibration Photometer Worksheet

Procedu	ure & Section Date & Time	564420	18-45	Instrument Var. Atten.	MRI
Liabtir	ng Conditions			var. Alleri.	
•	Γ Photometer	Lights out	in high bay		
Readings-					
neadings					
9*					
Aver	age Reading				
Labsphere	Photometer				
Readings		10-8			
	1-187	(181	6171	6169	616
	1.168	1.175	1-176	1184	1.163
Aver	age Reading	, v			
	Comments				
		AMBU	5NT-Lesh	tsoutinh	Jh buy

0 0	onditions otometer	2/27/03 Lamp- W/A-	20:00 high hay lan	Var. Atten.	VA 15
Average	Reading				
J	.		•		
Labsphere Pho	otometer				
Readings	-	× 15-8			
	502	4505	11482	4507	LA93
1.	49	6482	1.506	1.498	1.485
Average	Reading				•
Co	mments				
		post	-30° Fla	ats 15	3.61V nderral
Co	omments [DOG+	-300 Fl	atc 15	3.61V

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Operator:

Instrument Platform Calibration Photometer Worksheet

Lightir	ure & Section Date & Time ng Conditions T Photometer	Mar 2 200:	8-34 3 20:21 Indow NOU	Instrument Var. Atten.	MRI 6
Readings					
•					
Aver	age Reading				
Labsphere Readings	Photometer	X10-4			
_	1.227	1,240	1,216	1.237	1,243
	1,221	1-233	1.213	1:229	1.231
Aver	age Reading Comments	Post fol	dr 1 x10	-8	,
		1-201	-201 (-201 (-221 1.196	•	.198 .200

Lightir		564420 S Mar 2 2003 Lamp +		Var. Atten.	
ricadings				I	
Ave	rage Reading	I			
Labsphere	e Photometer				
Readings		XIO	-8	-	
•	1.536	1.542	1.555	1.537	1.525
	1.543	1,577	1,526	11427	1:53
Aver	age Reading				
	Comments	X10-8	•		
Feller6 Seepe	Between model allmodes	1,542 1.5 1,567 1.	57 1.55 57 1.5	9 154 71 1.558	1.541



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Instrument Platform Calibration Photometer Worksheet

Procedure & Section Date & Time Lighting Conditions Labsare UDT Photometer	17:45 N	1er 03 103 external t	Instrument Var. Atten.	MRI 255 NOUDT
`ReadingsX	10-6		7 2 2	
2-108	2-908	2-909	2.10+	2-907
2-908	2-908	2-908	2-908	2-908
Average Reading	19:50	•		
` Labsphere Photometer				
Readings	X10-6			
2-909	2-909	2-909	2.909	7-909
2910	2-910	2911	2.911	2-912
Average Reading				
Comments		•		
				İ

564420 MRI Instrument Procedure & Section Mer 3 563 Var. Atten. Date & Time 20:40 Lighting Conditions no UDT Laborh UDT Photometer POST Filter 8 Readings X10-6 2-914 2-913 2-913 29.4 2914 2-913 2913 Average Reading 21:30 glerne Gelter 7 Labsphere Photometer X105 Readings 2916 2-916 2-9,6 2915 7916 2-916 2.916 2-916 2916 Average Reading VA=256 21133 Comments Ambient (-006 1-007 1006 X10-6 1.066 1-008 1-352×10-8 1.326 1-007 1-066 1-006 1-007 1.006 1-336 1338 3-541 3-551 3.552 3.55 3.546

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Instrument Platform Calibration Photometer Worksheet

Procedure & Section Date & Time Lighting Conditions	Mar 3 103	21:50	Instrument Var. Atten.	
≺UDT Photomete r				
Readings 文 10	-8			
4.539	4-536	4.52	4,513	4,549
4,521	4,530	4,515	4,533	4.534
Average Reading			•	,
Post Subr2				0.5
Labsphere Photometer		UA = 42	time =	2455
Readings メリ	0-8	•		,
4-453	4.465	4.449	4.459	4.48
4,441	4.46	4.462	4.444	4,456
Average Reading		_		:45
Comments	X10-8		time 23	, 45
Selfer 3	4-388	4401	4.40 4-39	1402
VA=42	4-38	4-377	4-392 4-39	19 4-409

Post

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Procedure & Section	564420	Instrument <u>MRF</u>
Date & Time	Mar 3 2003 23:50	Var. Atten. 142
Luby Lighting Conditions -UDT Photometer	Larp twindow rol	
	0	
Readings <u>VA=0</u>	Ambient 410-8	
1.194	1,190 1,175	1202 1.183
1.191	1-203 1-197	1.173 1.178
Average Reading		MOUPT
Labsphere Photometer	Post F-lfr4 La	n, VA=25
Readings		
2.269	2-285 2-264	2-262 2-283
- / -	2-262 2-273	2.267 2292
Average Reading		alim
Comments	POST Felber 5 Lang	,twinder NOODI U
XIV-8	2-28 2-298 2-29	, 2-248 2-285
•	2-258 2-258 2-2	sturder NOUDT 01:00 2248 2-285 74 2-284 2-253
,X(6-8		
VA-20 Ambient	1.185 1.192 1.189 1.179 1.190 (.203	1.188 1.173 3 1.181 1.201
00	SIZE CAGE CO	DDE DWG. NO.

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Date	Lamp	Time On	Time Off	Chron On	Chron Off	Current (A)	Voltage (V)
26-Feb-03	D	10:45	11:02	N/A	N/A	N/A	N/A
26-Feb-03	O	11:28	13:03	10:05:49	11:35:07	6.25	N/A
27-Feb-03	C	8:30	11:25	11:35:07	14:25:01	6.25	13.64
27-Feb-03	С	17:30	20:11	14:25:01	17:06:50	6.25	13.61
2-Mar-03	С	17:00	23:45	17:06:50	23:49:59	6.25	13.44
3-Mar-03	С	16:20	16:57	23:49:59	24:27:00	6.25	13.44
3-Mar-03	Α	16:42	17:01	4:50:00	5:09:00	N/A	N/A
3-Mar-03	С	16:58	1:00	24:27:00	32:23:32	6.25	13.61
3-Mar-03	Α	16:59	21:30	5:04:00	9:30:34	N/A	7.81
3-Mar-03	D	17:03	21:31	6:00:18	10:24:36	N/A	8.28